

INVESTIGATION INTO THE GROUND AND FLIGHT CHARACTERISTICS OF J-100 AND J9-10-300 NEOPRENE BALLOONS

FINAL REPORT

COVERING PERIOD OF JUNE 1, 1957 - JANUARY 1, 1958

RLP#

DEWEY AND ALMY CHEMICAL COMPANY Division of W. R. Grace & Co. CAMBRIDGE, MASSACHUSETTS

DEWEY	AND	ALMY	CHEMICAL	COMPANY	RESEARCH	LABORATORY
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PROBLEM #						

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OBJECT OF RESEARCH

STUDY THE GROUND AND FLIGHT CHARACTERISTICS OF THE DAREX J-100 AND J9-10-300 BALLOONS CARRYING CERTAIN SPECIFIED PAYLOADS AND FREE LIFTS.

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	REPORTED BY:
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I. ABSTRACT

An account is given of a series of ground and flight tests designed to impart information regarding the performance characteristics of the J-100 and J9-10-300 DAREX neoprene ballons.

The following parameters were considered:

- 1.) Burst diameter
- 2.) Altitude
- 3.) Rate of rise
- 4.) Permeability to hydrogen at sea level
- 5.) Uniformity of performance
- 6.) Adverse weather flight characteristics
- 7.) Aging characteristics

The capabilities of the J-100 ballon in particular have been extensively examined and, wherever possible, explanations have been given for the phenomena encountered during the test program.

It was found that:

- 1.) Because of the influence of flight time and temperature on the elongation capabilities of the ballons, those balloons having a low ascensional rate did not achieve as large a burst volume as those having a high ascensional rate. Consequently, the bursting altitude for a given payload was independent of the free lift in the range considered.
- 2.) Because of the inverse relationship between ground volume and burst altitude, the altitude decreased progressively with each increase in payload.
- 3.) The ascent rate for J-100 balloons in the range considered ranges from 250 feet per minute with 50 grams of free lift to 660 feet per minute with 500 grams of free lift. For the J-9-10-300 balloons the ascent rate ranged from 500 feet per minute with 300 grams of free lift to

850 feet per minute with 1000 grams of free lift.

- 4.) The permeability to hydrogen at sea level was found to increase in direct proportion to the gross lift. The rate of loss of lift was found to be 11.4 grams/hr. for a 1 lb. load and increased progressively to 48.4 grams/hr. for a 6 lb. load.
- 5.) The uniformity of performance of the J-100 balloons was found to be such that the average standard deviation for a group of balloons having the same payload and free lift is \neq 2,600 feet. The ascensional rates within a group having the same payload and free lift were generally within \neq 10% of the average.
- 6.) It was observed that overcast skies did not appreciably affect the bursting altitude of the J-100 balloons.
- 7.) Aged balloons performed well up to a period of 1 year of shelf-storage.

II. INTRODUCTION

This report covers a series of tests designed to impart more precise technical and operational information on Darex J-100 and J9-10-300 balloons than is currently available. More specifically, it is a study of the ground and flight characteristics of these balloons for their application to carrying unusual and varied payload and free-lift combinations. As the science of meteorology advances, it becomes desirable to have a greater control and better understanding of the instrument-carrying balloons used either in studying the atmosphere or in other wind-dependent operations. The J-100 balloon heretofore has been used primarily as a pilot balloon to determine wind direction and velocity at various altitudes; also the height, direction and velocity of clouds whenever the balloons enter their bases. Little is known concerning the ability of these small balloons to carry unusual payloads or of their ascensional rates with low free lifts. This study is aimed at clearing up some of the questions involved in this respect, in order to enable the accurate projection of a predetermined payload to a predetermined point in space at a known rate of ascent.

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III GENERAL DISCUSSION

- A. Manufacture
- B. Theoretical Considerations

III. GENERAL DISCUSSION

A. Manufacture

Darex meteorological balloons are manufactured from a specially prepared duPont neoprene latex, compounded by the Dewey and Almy Chemical Company to impart a high degree of stretch, cold resistance, gas impermeability, and resistance to deterioration due to aging and exposure. The process used in manufacture is the single-compound-dip, gel-expansion method, wherein an impervious mold is coated with coagulant and dipped into the latex compound causing a thin gel to form on the mold. Progressive diffusion of the coagulant salt through this thin layer coagulates to form a thicker, finely knit gel. The thickness of the gel is determined by the length of time the mold is allowed to dwell in the compound. Once the desired thickness is obtained, the mold is removed from the compound and allowed to air set for a short time until the process of spontaneous exudation of serum, known as synaeresis, commences. The gel is then immersed in water which causes an osmotic flow of the serum to take place with a consequent increase in total solids. The gel, after toughening, is stripped from the mold, washed, and inflated to 4.75 times its original diameter and dried to permanently increase its size. The mechanism by which it is believed this permanent increase in size takes place, is that during the expansion of the gel the polymer micelles move on their water matrices until such time as when the area has become so large that the now thinly dispersed water no longer affords lubrication and the micelles come in contact with each other resulting in adhesion and loss of further plastic movement. Inflation beyond this point becomes elastic and reversible. Once the balloon is dried it is vulcanized in a hot air oven for six hours at 230° F. then inspected and packed in cartons to be opened again only when ready for use.

The factory specifications for the balloons studied in this project are the following:

			Sea-Level		Thic	kness
Type	Weight	Diameter	Burst Diameter	Sea-Level Burst Volume		At Burst
J-100	100 gms.	16 inches	90() inches	216 cu. ft.()	.004 inch	.0001 inch
J9-10-300	300 gms.	29 inches	12.5 () feet	1,023 cu. ft.().004 inch	.0001 inch
B. Theor	retical Con	siderations	<u>3</u>			

Archimedes law states that the buoyant or upward force exerted on a body immersed in a fluid is equal to the weight of the fluid displaced. A body immersed in fluid displaces its own volume of fluid. If the weight of the fluid displaced equals the weight of the body, the body is in equilibrium. If the weight of the fluid displaced is greater than the weight of the body, the body rises, and if the weight of the fluid displaced is less than the weight of the body the body falls. Thus if a balloon is filled with a gas having a density less than that of air, the balloon will possess a buoyancy or gross lift as expressed by the following equation:

$$L = V(d_a - d_g)$$
 where: $L = gross \ lift \ or \ buoyancy, = free \ lift + payload + weight \ of \ balloon$

$$V = \ Volume \ of \ gas.$$

$$d_a = Density \ of \ air.$$

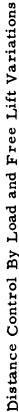
$$d_{\sigma} = Density \ of \ gas.$$

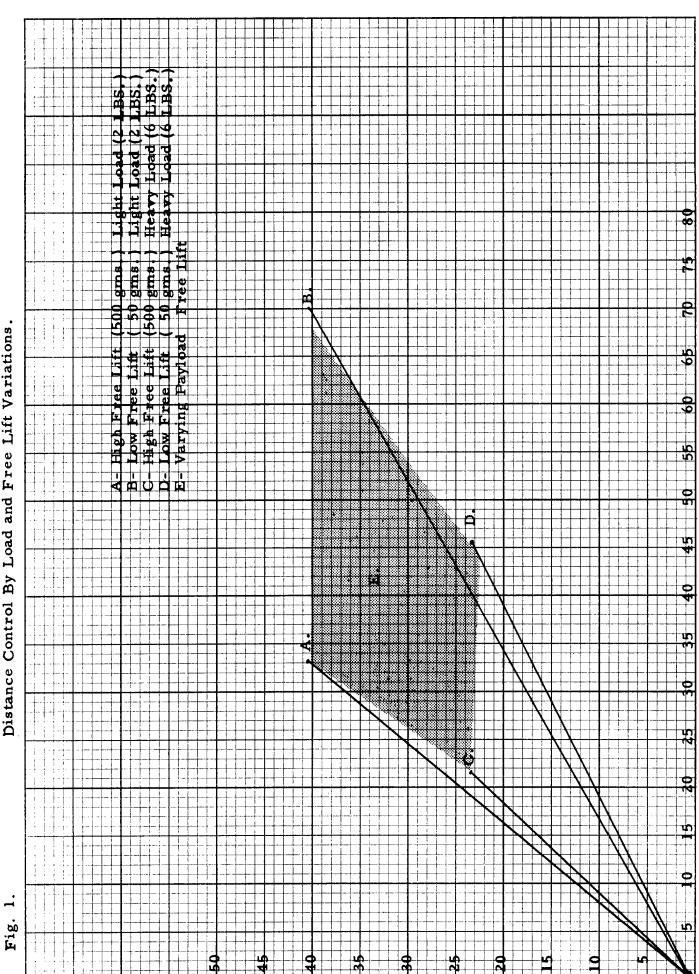
In the case of an extensible balloon, if the gross lift is greater than the weight of the balloon and its payload, (i.e. if the balloon possesses a "free-lift"), and if the balloon is tied off so that no gas can escape, it will rise. As it gains altitude the atmospheric pressure decreases because of a decrease in air density. This decrease in air pressure allows the volume of the gas inside the balloon to increase and thus expand the balloon. The balloon continues to rise

and expand in this manner until it reaches a point where the film becomes so thin that further expansion is no longer possible and rupture occurs, thus terminating the flight. The relationship between gross lift and altitude can easily be seen if we can assume a most probable bursting volume. If a balloon starts its ascension with a large ground volume it will require a certain decrease in atmospheric pressure to expand to its bursting volume. A similar balloon starting off with a smaller volume will require a greater decrease in pressure and thus a higher altitude, to arrive at its bursting volume. Since the ground volume depends on gross lift it is evident that altitude is also dependent on gross lift.

Thus it is possible to control the altitude at which a neoprene balloon will burst by choosing the proper payload and free lift. Since the free lift also determines the ascensional rate the time spent in flight can also be controlled. This relationship is represented graphically in figure 1.

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30 M.P.H

Mean Wind Speed of

Assuming A

Ground Distance (Miles)

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IV. GROUND AND FLIGHT TEST DATA AND ANALYSIS

- A. Test Program
- B. Theoretical Relationship Between Ground Diameter, Burst Diameter and Altitude.
- C. Ground Burst Diameters versus Flight-Test Burst Diameters.

A. TEST PROGRAM

In order to obtain a more accurate knowledge of the capabilities of the J-100 and J9-10-300 Darex neoprene balloons the following test program was carried out during this project.

TASK I. Existing Data and Ground Tests

Phase (a) Study the existing data on J-100 balloons as compiled by the U. S. Weather Bureau. The balloons tested by the Weather Bureau carry no payload, bursting elevation being determined via theodolite tracking technique, and are valuable in determining the maximum altitude which can be obtained by the J-100 balloon.

Phase (b) Conduct a series of ground burst tests of the J-100 and J9-10-300 balloons, using existing laboratory facilities, to determine what bursting dimensions can be normally expected at sea level. These results are to be compared with a series of flight tests in order to ascertain whether or not the balloons are reaching their maximum burst diameters during flight.

Phase (c) Investigate variations in free lift (and consequently gross lift), caused by adiabatic cooling of inflation gas at various rates of flow. Since the volume of a gas varies directly with the temperature of the gas (at constant pressure), more gas than is necessary may be put into the balloon to achieve a given free lift. As the gas becomes warmer it may turn out that the balloon has more lift than is intended.

Phase (d) Plot the room temperature diffusion rate of the J-100 balloon inflated to flight dimensions. Sometimes it becomes desirable or even necessary to hold an inflated balloon for a period of time before launching. In these cases a knowledge of the amount of diffusion, and consequently lifting force lost, becomes valuable.

TASK II. Study the performance of the J-100 and J9-10-300 by Field Testing.

A series of flight tests were carried out under varying weather conditions

and varying payload-free lift combinations. Altitudes and ascensional rates of the balloons were determined to a high degree of accuracy by use of radiosonde tracking technique. Since the standard radiosonde weighs in excess of two pounds, a problem was encountered in those flights which required a payload of exactly two pounds. This problem was overcome by removing the radiosonde sensor mechanism from its regular plastic housing and re-installing it in a hand-made housing of lightweight rigid styro-foam. (For details of procedure see Figures 2A, 2B.) The flight test program was broken down into the following phases:

Phase 1.

J-100 clear weather daytime flights. External loads of 2, 3, 4, and 6 lbs. Free lifts of 50, 100, 200, 350, and 500 grams for 2 3 lb. payload, 200 500 grams free lift for 4 6 lb. payload. Five flights for each payload-free lift combination making a total of 70 flights in this phase.

Phase 2.

J-100 adverse weather daytime flights. External load of 3 lbs. Free lift of 200 grams.

Weather Conditions:

Light cloud cover - 5 flights

Heavy cloud cover - 5 flights

Rain - 5 flights

Total of 15 flights in phase 2.

Phase 3.

J-100 aged balloons, clear weather, daytime flights. External load of 3 lbs. Free lift of 200 grams. Total 5 - 20 flights. Enough flights were flown on aged balloons to determine at what point during shelf-storage their extensibilities become impaired.

Phase 4.

J-100 nighttime flights. External load of 3 lbs. Free lift of 200 grams.

Weather Conditions:

Clear - 10 flights

Light cloud cover -)

- 5 flights

Heavy cloud cover-)

Phase 5.

J9-10-300 clear weather daytime flights. External loads of 5 and 10 lbs. Free lift 300, 700, and 1000 grams for 10 lb. load. Free lift of 300 and 700 grams for 5 lb. load, 4 flights for each payload free lift combination.

Phase 6.

J-100 clear weather daytime flights. Balloons preheated by a treatment of five minutes in water at a temperature of 180°F. minimum. External load of 3 lbs. Free lift of 200 grams. Fresh balloons, (no more than 3 months old,) are used in this phase to determine whether any crystallization has taken place between time of manufacture and time of use.

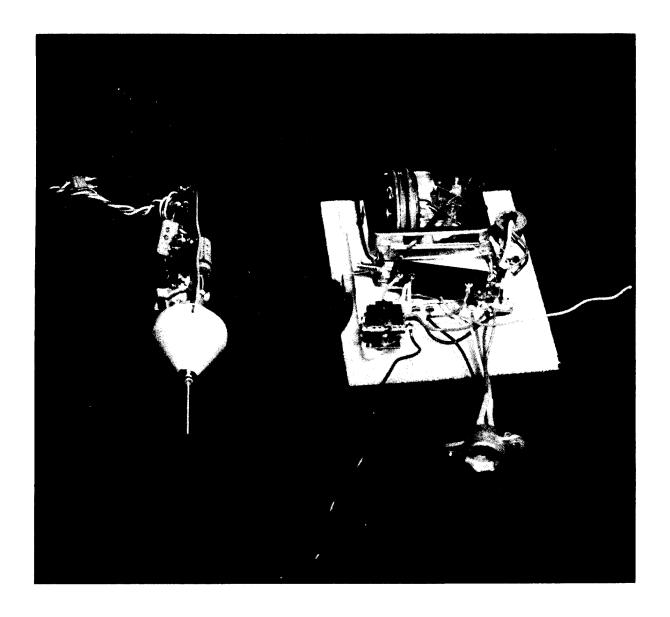
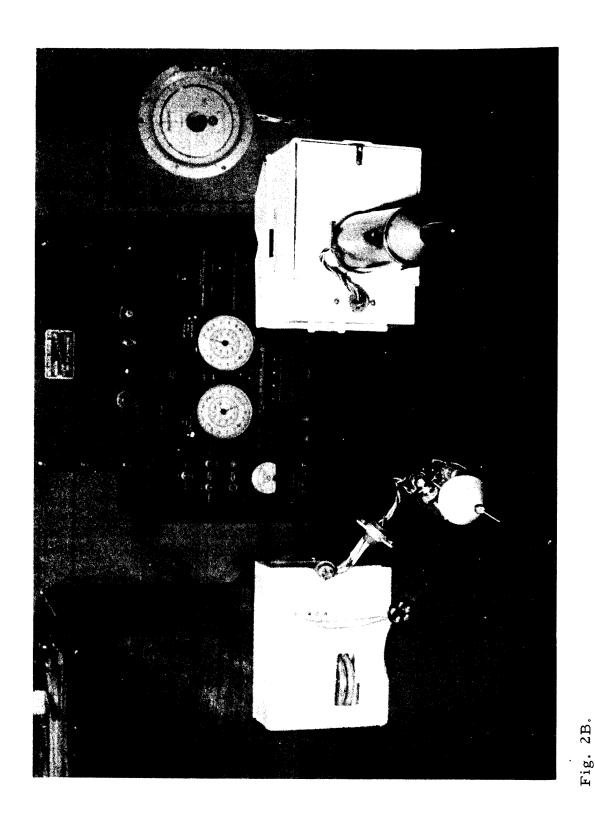


Fig. 2A.

Sensor mechanism is stripped of its plastic housing.

Transmitter is removed from its casing.



Sensor mechanism is installed in light-weight styro-foam housing (right). Original size of standard radiosonde is shown at left.

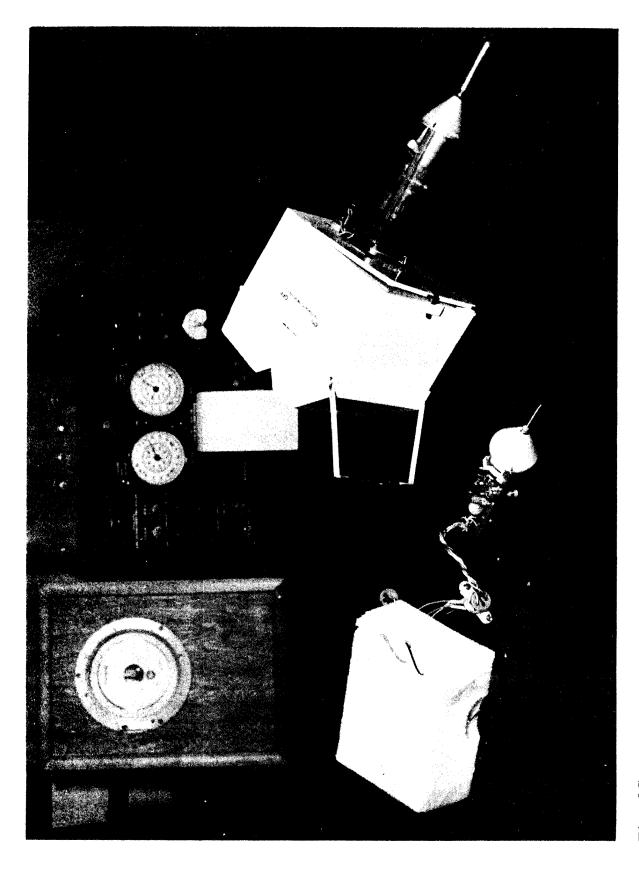


Fig. 2C.

Procedure for modifying radiosonde:

A standard AN/AMT-4A radiosonde is stripped of its white plastic housing as follows:

- 1.) Remove the studs which anchor the sensor mechanism platform to the housing.
- 2.) Cut the wire leading to the humidity element.
- 3.) Cut the wires leading to the metal thermistor clamps.
- 4.) Remove the sensor mechanism from the housing.
- 5.) Prepare light-weight styrofoam housing by cutting a total of 6 sections to the following dimensions; 2 sections 6.25x4.75 inches (these will be the top and bottom of the new housing); 2 sections 6.25x2.25 inches (these will be the sides); 2 sections 4.5x2.25 inches (these will be the front and back).
- battery wire connections. (fig. 2B)
- 7.) Cut away part of the bottom to prevent housing from touching aneroid gettows (fig. 2B)
- 8.) Make sure transmitter cable and ground-check plug are outside of housing. (fig. 2B)
- 9.) Build up sides and top of housing with the aid of scotch tape.
- 10.) Cut two small holes in top of housing and expose temperature circuit wires. (fig. 2C)
- 11.) Attach thermistor to temperature leads. Plug in transmitter.
- 12.) Plug in battery and tape battery to the housing.
- 13.) Cut away part of side to expose on-off switch. (fig. 2C)
- 14.) Radiosonde is now ready for use. No humidity element is used.
- 15.) Remove transmitter housing and cover transmitter with transparent polyethylene bag if further weight reduction is desired.

B. Theoretical Relationship Between Burst Volume and Altitude

The volume of a balloon at burst is calculated from the relationship:

$$V_2 = NR \quad \frac{T_2}{P_2}$$

where: V_2 = volume at burst (cubic feet)

K = gross lift (lbs.) ÷ (molecular wt. of air - molecular wt. of H₂)

 $R = 1.33 \times 10^3$

T₂ = temperature at burst (%)

P₂ = pressure at burst (millibars)

Thus, if a most probable burst volume is assumed from a series of ground burst tests, the theoretical bursting altitude can be predicted from the relationship:

$$\frac{\mathbf{T}_2}{\mathbf{P}_2} = \frac{\mathbf{V}_2}{\mathbf{NR}}$$

For the sake of convenience in predicting bursting altitude when the most probable burst volume is known, a graph (fig.3) has been prepared wherein altitude is plotted as a function of $\frac{T}{P}$. The atmospheric conditions on which the graph is based have been derived from soundings taken at Vernalis, California, the site where the flight tests in Task II were carried out.

Example: Ground burst tests indicate that J-100 balloons will burst at an average volume of 268 cubic feet. Thus for a ballon carrying a 3 lb. load with 200 grams of free lift:

$$N = 3.67 \text{ lbs./26.9} = .136 \# \text{ moles}$$

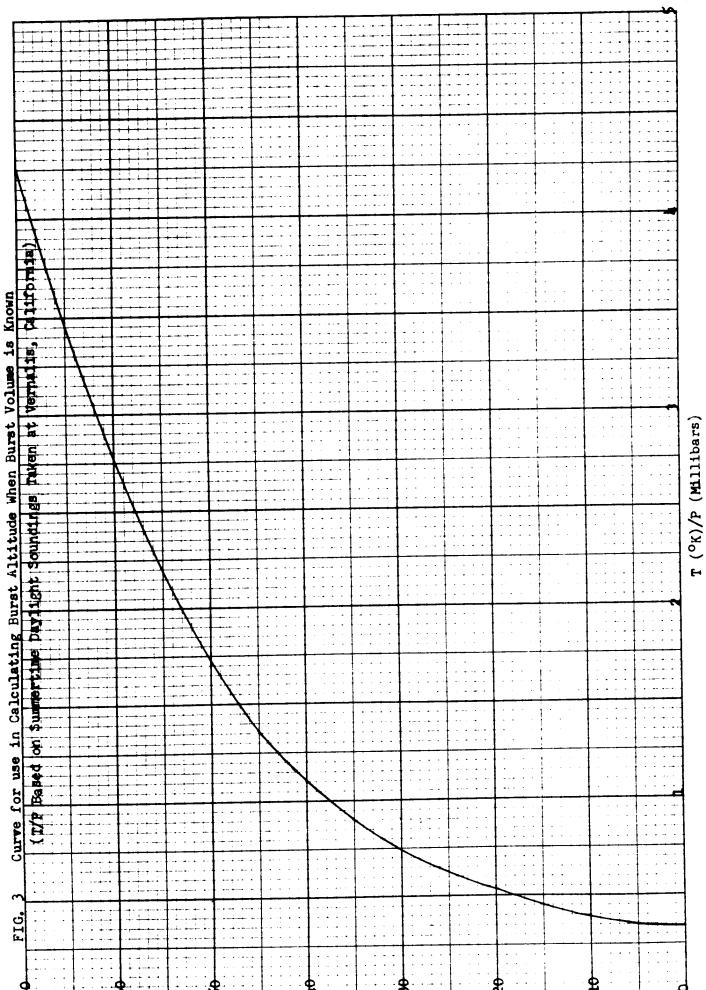
$$\frac{268}{(.136)(1.33 \times 10^3)} = 1.47 = \frac{T_2}{P_2}$$

Referring to the graph in fig. 3 we find the altitude to be 46,500 feet.

Using the data from the ground burst tests and referring to fig. 3,

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the expected performances of the J-100 and J9-10-300 balloons were calculated and compared to actual flight tests in table 2.



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TABLE 2.) EXPECTED VERSUS OBSERVED BURST ALTITUDE

J-100 Balloons

Average Ground Burst Volume = 268 cubic feet = 96 inches Burst Diameter Observed Observed						Diameter Observed
Pavload	Free Lift	N	Theoretical	Observed Altitude	Burst Volume	Burst Diameter
(lbs.)	(grams)	(# Moles)	(feet)	(feet)	(feet ³)	(inches)
2	5 0	.0884	56,000	40,930	132	75.5
2	100	.0925	55,000	38,809	131	75.3
2	200	.1010	53,000	40,636	155	79.6
2	<i>35</i> 0	.1130	50,800	41,190	179	83.8
2	500	.1251	48,500	39,018	174	82.9
3	<i>5</i> 0	.1251	48,500	33,666	143	77.4
3	100	.1301	47,800	36,495	165	81.3
3	200	.1370	46,500	35 ,3 12	168	81.8
3	350	.1502	44,900	35,285	181	83.9
3	50 0	.1625	42,900	34,736	191	85.4
4	200	.1750	41,000	30,736	180	84.1
4	50 0	.1988	37,500	29,900	194	85.9
6	200	-2499	32,000	23,741	202	86.8
5.75	620	.2740	29,200	23,000	218	89.3

J9-10-300 Balloons

A	verage	Ground	Burst Volume =	1940 cubic	feet = 185	inches	Burst Diameter
5	i	3 00	.2398	80,000	58,187	803	138
5	i	1000	.2960	76,000	57,565	993	147
10)	300	.4240	64,900	45,392	781	137
10)	700	.4580	63,000	46 ,2 12	876	142
10)	1000	.4810	62,000	46,220	930	144

The calculation of the exact bursting volume of a balloon in flight presents a problem. It is well known that the temperature of the gas inside the balloon during a daytime flight is substantially higher than that of the surrounding air. During a night flight, the temperature of the gas is equal to or slightly colder than the outside air. Therefore, in order to calculate the exact volume of a balloon in flight from the relationship PV = NRT, it is necessary that we know the exact temperature of the gas inside the balloon.

Some experiments designed to measure the internal temperatures of balloons in flight have been carried out by the National Bureau of Standards which indicate that the internal temperature of balloons in the altitude range under our consideration is as follows:

Day Flight Regular Balloon*

Altitude (ft.)	External Temperature (°C)	Internal Temper Top of Balloon	ature (°C) Bottom of Balloon
20,000	-22.5	-20.0	-10.0
25,000	-28.0	- 25.0	-19.0
30,000	-43	-37.0	-27.0
35,000	-5 2	-45	-36
40,000	– 60	-48	-4 2

For night flights the Bureau of Standards reports internal temperatures as ranging from 0°C to 6°C colder than external temperatures. These flights were all carried out during the months of December and January. Presumably, flights carried out during the summer would give slightly different results. It is also suspected that flights carried out at different time of day would differ from each other. Also, it can be reasonably assumed that the internal temperature of a balloon with a low ascensional rate would differ from that of a balloon with a high ascensional rate since the faster

balloon spends less time exposed to the sun and is better ventilated than the slow one.

A further complication arises if one considers the variance in internal pressure which must exist. A cold neoprene balloon skin exerts more internal pressure than a warm one. Presumably, balloons flown at night possess a higher internal pressure than those flown during the day, and those flown during the day differ from each other to a degree dependent upon the amount of heating received by the film. Thus far, to our knowledge, no one has been able to measure accurately the temperature and tension of a balloon film in flight.

Thus, in the absence of precise experimental data, no attempt was made to measure the exact bursting volumes of the balloons flown during this program. Calculations were simply based on the external temperature and pressure as reported by the attached radiosonde.

^{*}National Bureau of Standards Report #2530, June 12, 1953, Technical Report No. 8 "Measurement of The Temperatures Inside and Outside of Sounding Balloons During Flight"

As can be seen from Table 2, the burst diameters and, consequently the elevations attained in actual flight were considerably below those predicted on the basis of ground burst tests. Moreover, the burst diameters tend to increase as the free lift is increased and contrary to expectations, reach their greatest dimensions with the heavier rather than with the lighter loads. This phenomenon is believed to be largely a function of time and temperature. The following reasons are offered in support of this belief.

1.) EFFECT OF TIME AND TEMPERATURE

In previous low temperature testing of balloon films by the Dewey and Almy Balloon Laboratory (1) it was found that at relatively high temperatures the elongation of rubber like materials at rupture was essentially constant and relatively high, i. e., the material was soft and elastic. At very low temperatures the elongation was constant but very low, i.e., the material was hard and inelastic. In the range between the relatively high and very low temperatures the elongation decreased with the lowering of the temperature, i. e., the elastomer progressively underwent a change from a soft elastic material to a hard inelastic material. (See Table 3.) The sequence of events taking place during this phenomenon is believed to be as follows: The expansion of long chain polymers exhibiting rubber like properties is the result of two distinct movements; one the micro-movement of one molecule in relation to an adjacent molecule and the other the macro-movement of a group of molecules or a micelle in relation to another. These two movements occur simultaneously during expansion at normal temperatures. With a decrease in temperature, the macro-movement of the rubber micelles is retarded allowing greater opportunity for strain between the molecules. At this point there is a rise in modulus of the film as the polymers resist the forces tending to deform them. Consequently, some spot on the film, weaker than the rest, gives way and rupture occurs. The rise in modulus of the

film is attributed to the phenomenon known as crystallization wherein the movements of the carbon atoms in the polymer chain are restricted.

The relationship between extensibility and crystallization can be represented by drawing an analogy between a polymer chain and a coiled steel spring. If the spring is stretched out until all the coils are straightened, it will have a certain length. But, if we solder a few of the coils together so they cannot straighten out, then we will be unable to draw out the spring to its original length. Thus, if some of the polymers which go to make up a balloon film are prevented from uncoiling, because they are crystallized, the balloon will not achieve its maximum elongation. This, we believe, is the reason why balloons in flight do not achieve the same elongations as balloons inflated at room temperature.

Research by others (9) into the mechanism of high polymer crystallization indicates that the development of crystallinity in polymers is not instantaneous at low temperatures but progresses with time of exposure.

Curves of specific volume of rubber as a function of time at temperatures below the crystalline melting point show that crystallization cannot be considered complete for long periods of time. Thus, the maximum elongation which can be attained by a balloon is inversely proportional to the time of exposure to cold temperatures.

TABLE 3.) COLD CABINET TESTS - PROTOTYPE NEOPRENE BALLOONS

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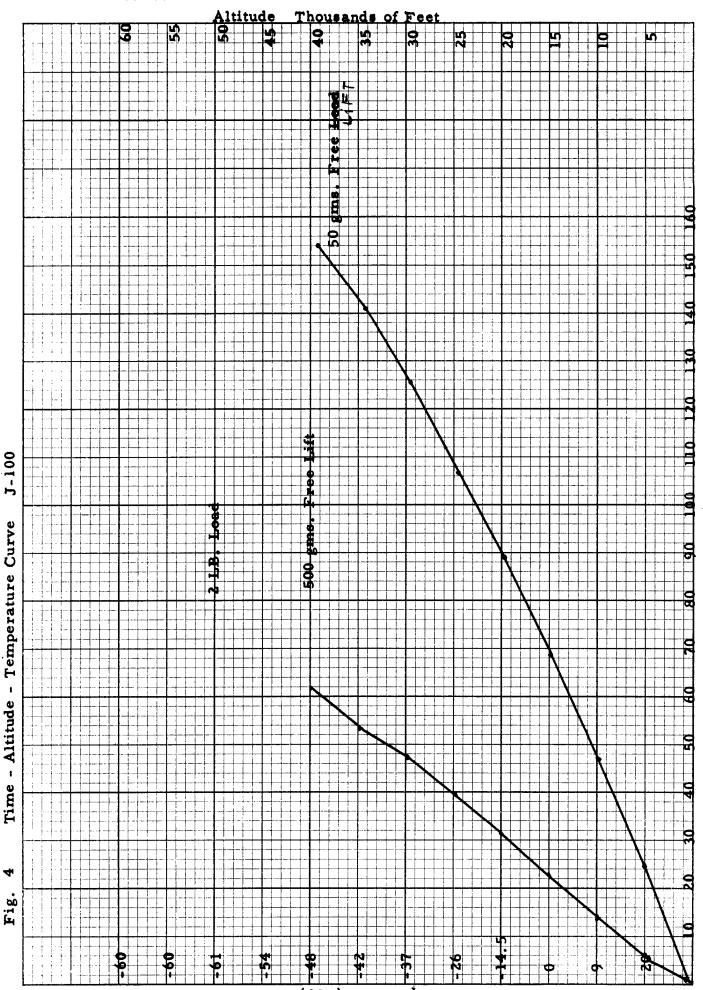
BALLOON #	101	102	103	113	114	115	116
GAUGE THICKNESS (IN.)	.0040	.0042	.0040	. 0045	.0040	.0045	. 0045
FLACCID DIAMETER (IN.)	4.63	4.25	4.70	4.50	4.50	4.375	4.56
MAXIMUM PRESSURE (Inches H ₂ 0)	4.8	5.1	3.0	14.7	15.6	23.0	23.0
PRESSURE AT 10 INDIA.	3.0	3.5	2.1	4.7	4.7	22.0	22.0
" " 15 "	2.1	2.6	1.8	4.9	5.4	22.5	21.0
r r 20 11	1.9	2.1	1.4	10.3	12.8		
" " 25 "	2.0	2.8	1.6				
" " 30 "							
PRESSURE AT BURST (In. H ₂ 0)	2.0	2.8	2.5	14.7	15.6	23.0	19.4
DIAMETER AT BURST (IN.)	25.5	25.0	29.5	21.9	21.5	18.5	19.5
TEMPERATURE °F.	72 ⁰ F	72 ⁰ F	72 ⁰ F	-45°F	-45 ⁰ F	-76°F	-76°F
RATIO B. D. /F. D.	5.52	5.89	6.21	4.87	4.78	4.23	4.28

Note that the internal pressure of the balloon rises sharply as the temperature decreases, indicating stiffening of the film. Since the area of the balloon varies as the square and the volume as the cube of the diameter, the normal tendency is for the pressure to decrease progressively with increase in diameter although there is actually an increase in linear stress in the film. However, as the balloon stiffens as a result of the low temperature, a much greater pressure is developed per unit increase in surface area.

TABLE 4.) PAYLOAD VS. TEMPERATURE AT BURST

J-100 Balloons

Payload	Free Lift	Average Temperature at Burst	Average <u>Diameter at Burst</u>
2 lbs.	200 grams	-48°C (-54.4°F)	79.6 inches
3 lbs.	200 grams	-43°C (-45.4°F)	81.8 inches
4 lbs.	200 grams	-36°C (-32.8°F)	84.1 inches
6 lbs.	200 grams	-19°C (- 2°F)	86.8 inches
•		J9-10-300 Balloons	
5 lbs.	300 grams	-62°C (~79.6°F)	138 inches
5 lbs.	1000 grams	-56°C (-68.8°F)	147 inches
10 lbs.	300 grams	-59°C (-74.2°F)	137 inches
10 lbs.	700 grams	-60°C (-76.0°F)	143 inches
10 lbs.	1000 grams	-56°C (-68.8°F)	144 inches



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TO DIVISIONS PER INCH BOTH WAYS

This temperature effect is borne out in actual practice by the fact that in the case of the J-100's those balloons carrying a six pound payload attained a larger diameter than those with a two pound payload. This is in agreement with the concept of temperature effect since balloons carrying the heavier loads reach their maximum diameter at an altitude far below, and consequently much warmer than balloons carrying two pound loads. (See Table 4.) In this connection, it is interesting to note that in the case of the larger (J9-10-300) balloons the greatest burst diameters were reached by those having the highest free lift, but there was no substantial difference in burst diameters between heavy and light payloads. The "payload effect" did not appear in this case since the maxima of the larger balloons is such that very low temperatures are encountered even in the case of the balloons carrying ten pound loads.

The fact that balloons with the highest free lift attained larger burst diameters than those with the lower free lift and similar payload, can be accounted for by the fact that those balloons with a low free lift spend a considerably longer time, (See Fig. 4.) in regions of low temperature than those with high free lift, therefore allowing a greater degree of crystallization to take place with a resulting decrease in extensibility.

2.) EFFECT OF PAYLOAD:

It would seem, at first thought, that balloons in flight should burst at a smaller diameter than balloons tested on the ground, because of the strain induced on the balloon film by the payload. On the other hand the flight tests appear to indicate that the higher the payload, the larger the burst diameter. This seeming paradox was investigated via two methods: 1.) Study data on balloons flown with no payload attached. 2.) Perform experiments whereby various loads are attached to balloons burst at sea level.

Phase 1.

The U.S. Weather Bureau, as a matter of routine, carries out flight tests on production lots of Darex J-100 balloons. In these tests, the balloons carry

no payload and have a free lift of 500 grams. Burst altitude is determined via the odolite tracking technique. The Weather Bureau was contacted and the following flight test data were received:

TABLE 5.) FLIGHT TESTS WITH NO LOAD ATTACHED

J-100 WHITE BALLOONS

RELEASE			
DATE 1956	TIME I	OT 615	
	<u> </u>	URST ALTITUDE (FEET)	BURST DIAMETER (IN.)
5/27	0730	49, 4 75	74.0
5/28	0730	56,975	85.0
5/30	0730	59.950	91.0
6/3	0730	55,050	83.5
6/7	0730	59,950	91.0
6/9	0730	56, 975	85.0
6/10	0731	59,950	91.0
6/11	0731	54,230	81.0
6/16	0730	<u>55,050</u>	83.5
	AVERA	GE 56,300	85.0

As can be seen, even balloons carrying no payload burst at a substantially lower diameter during flight than those burst at sea-level. It is also noted that these balloons burst at approximately the same diameter as those flown with a payload and the same free lift (see 500 grams free lift; 2, 3, 4, 6 lb. loads, Table 2.)

Phase 2.

To investigate further the effect of payload on burst diameter, an experiment was carried out whereby various loads were attached to a series of balloons burst at sea-level. The procedure was as follows: A J-100 balloon was fitted with a valve closure. A 2.1b, weight was suspended from the closure. The balloon was then inflated with helium until it just balanced the weight in mid-air. The

helium tank was then replaced with a compressed air inflation line and inflation was continued to burst thus maintaining a constant load on the balloon throughout the entire inflation. The test was repeated using a 4 and then a 6 pound load. Fifteen tests were carried out for each condition with diameters being measured at burst by means of moveable vertical rods. The results are as follows:

TABLE 6.) GROUND BURST TESTS WITH LOAD ATTACHED

LOAD	NONE	2 LBS.	4 LBS.	6 LBS.
BURST DIA.	95.0	9 4 . 0	96.0	85.5
(INCHES)	97.5	105.0	95.0	103.0
•	92.0	90.0	97.0	93.0
	92.0	100.0	95.0	84.0
	97.0	90.0	95.0	94.5
	100.0	90.0	94.0	850
	92.5	101.5	103.0	97.0
	93.0	105.5	97.0	87.0
	101.0	91.0	82.0	97.0
	94.0	86.5	90.0	95.0
	89.0	90.0	94.0	86.0
	92.0	96.0	94.0	92.0
	97.0	90.0	85.0	97.0
	94.5	90.0	94.0	93.0
	88.0	98.0	90.0	98.0
AVERAGE	94.5	94.5	93.5	92.2

It can be seen that although there is a definite downward trend with increasing load, the difference is not great enough to fully account for the comparatively low bursts achieved in flight. It appears therefore, that the strain induced into the balloon film by the attached load is only partially, if at all, responsible for smaller than expected burst diameters and definitely does not tend to increase burst diameters with increasing load. (It will be recalled that at first glance the flight test data seemed to indicate that the heavier the payload the larger were the burst diameters attained.) This further substantiated the theory of temperature effect.

3.) EXPOSURE TO OZONE:

Ozone is formed in the atmosphere by the action of ultra-violet light from the sun on molecules of atmospheric oxygen. While the amount of ozone varies somewhat with the seasons and in different parts of the country, the daily

concentration at the earth's surface is from 0 - 50 parts per 100 million parts air by volume. Previous work by the Dewey and Almy Chemical Co. Balloon
Laboratory revealed that neoprene balloons maintain most of their elongation properties at ozone concentrations up to 680 parts per 100 million parts air by volume (2). While it cannot be claimed conclusively that atmospheric ozone plays no part at all in decreasing the expandibility of neoprene balloons, it is considered to be a very small factor since concentrations larger than 50 parts per 100 million parts air are not usually encountered at the altitude reached by these balloons.

4.) ULTRA-VIOLET RADIATION:

The effect of ultra-violet radiation on neoprene balloons has been shown to be negligible by an experiment, performed by Barford et als (7), which made use of an "atmosphere chamber" to determine the effects of solar radiation on both natural rubber and neoprene type balloon fabrics. To simulate the conditions of a daytime flight, two large ultra-violet discharge tubes and twelve infra-red lamps were placed at distances in such a way that the total radiation was equivalent to that due to the sun in the upper atmosphere. The results of the experiment showed that natural rubber balloons are appreciably affected by solar radiation, but neoprene balloons are not affected at all.

BURSTING EXTENSION

<u>MATERIAL</u>	WITH RADIATION	WITHOUT RADIATION		
Natural Rubber	4.7	5.6		
Neoprene	4.2	4.3		

A further indication that solar radiation does not deteriorate the balloon fabric during flight is obtained when we compare night flight burst volumes to day flight burst volumes. Calculated on the basis of ambient temperature and pressure it can be seen from table 7 that there is close agreement between burst volumes. However, if we take into account the fact that the

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internal temperatures are higher during the day flights then it becomes evident that day flight balloons achieved a larger burst volume than those flown at night. Thus it can be reasoned that if sunlight did have a deteriorating effect on the balloon fabric then those balloons flown at night would have burst at a larger volume than those flown in the daytime.

TABLE 7.) NIGHT FLIGHTS VERSUS DAY FLIGHTS

J-100 Balloons

		Night F		Day Fl	
Payload	Free Lift	Altitude	Volume	Altitude	Volume
(Lbs.)	(Grams)	(Feet)	(Cubic Feet)	(Feet)	(Cubic Feet)
3	200	38,064	188	31,361	146
3	200	32,214	149	38,661	186
3	200	36,197	170	39,498	192
3	200	37,103	179	36,165	174
3	200	34,695	162	30,876	140
3	200	34,488	162		
3	200	37,795	179		
3	200	38,854	172		
3	200	35,804	167		
3	200	36,414	172		*********
	Average	35,9 63	170	35,312	168

5.) ATMOSPHERIC TURBULENCE EFFECT ON BURST DIAMETER:

It is not felt that atmospheric turbulence contributes to any large extent to lowering the extensibility of neoprene balloons, except perhaps during the initial phase of the ascent where, as the balloon rises against the dense air, a deformation of the upper hemisphere known as "dishing in" occurs. As a result of the internal restoring forces and the decrease in velocity, the balloon does not maintain its deformed shape and springs back to its original spherical shape. Once the balloon passes this stage and ascends into thinner and colder air, little or no deformation occurs since at the colder temperatures and higher elongation the balloon possesses greater rigidity.

C. PERFORMANCES OF AGED BALLOONS

It is well known that meteorological balloons, after years of shelf-storage, deteriorate to some extent. The term "aging" as applied to balloon films describes various physical and chemical changes that begin when manufacture is completed and continue during storage until the product is no longer suitable for its intended use.

The chemical changes in vulcanized rubber that take: place with aging are attributed to three possible reactions:

- 1. Chain scission the long molecular chains, which form the major structure of the polymer, may be cut into smaller pieces. This reduces the molecular weight of the polymer and, as molecular weight decreases, the tensile strength of the rubber is lowered and ultimately is lost completely. If "chain scission" occurs extensively in balloons, they will soften and appear to have become unvulcanized. This phenomenon is known to occur in natural rubber and is called "reversion". Neoprene, however, is very resistant to this type of degeneration and in this respect is superior to natural rubber as a balloon film.
- 2. Age Crystallization the linear molecular chains may be tied together by cross-links. The process of cross-linking due to aging is thought to be due to the effect of oxygen reacting with the unsaturated groups in the polymer. This process, if it occurs extensively, results in a decrease in extensibility and an increase in the stiffness of the film. In the case of a balloon, if the process of crystallization has not progressed too far, this condition may be reversed, through the action of heat, by immersing the balloon in hot water (200°F) for a few minutes prior to use.
- 3. The nature of chemical side groups along the molecular chains may

be modified. In case of neoprene, chlorine may be split off to form HCL.

These reactions have received considerable attention by many investigators with the results that compounding technique, have been improved to a point where any deterioration of a neoprene balloon film due to aging has been reduced to a very small minimum and has been mentioned above, can normally be reversed by "preheating!"

Since it is operationally desirable to use balloons without the necessity of preheating, a series of flight tests was carried out in an effort to determine at which point the J-100 balloons begin to lose their extensibility. The results of these tests are as follows.

TABLE 8.) AGED BALLOON FLIGHTS J-100 WHITE BALLOONS

LOAD	<u>LIFT</u> (<u>GRAMS</u>)	AGE	PREHEAT	AVERAGE ALTITUDE (FT)	AVERAGE BURST DIA. (INS.)
3	200	3 MOS	NONE o	35,312	81.8
3	200	3 MOS	5 MIN. @ 200 F	34,103	30 .4
3	200	13-15 MOS	NONE	32,518	79.3
3	200	19-20 MOS	NONE	31,367	78.1
3	200	24 MOS	NONE o	20,820	69.2
3	200	24 MOS	5 MIN. @ 200 F	31,227	77.5

The foregoing data indicates that the J-100 balloons maintain their meximum extensibility without the necessity for preheating for a period of time well over 12 months.

D. ADVERSE WEATHER FLIGHTS

For operational purposes it is essential to have a knowledge of the performance characteristics of balloons under adverse weather conditions for comparison with performance under clear weather conditions. This phase of the test program was designed to study the performance of J-100 balloons under the following adverse conditions: Heavy clouds, daytime; light clouds, daytime; rain, daytime; heavy clouds, nightime. The results of this series of tests is as follows:

TABLE 9.) ADVERSE WEATHER VS CLEAR WEATHER FLIGHTS

ALL FLIGHTS HAVE 3 LB. LOAD AND 200 GRAM FREE LIFT

TIME	WEATHER	ALTITUDE (FT.)
Day	Clear	35, 310
Day	Heavy Clouds	38, 731
Day	Light Clouds	36,004
Night	Clear	35,663
Night	Heavy Clouds	34,013

The data in TABLE 9 indicates that burst altitudes of J-100 balloons carrying the same payload and free lift are not affected by variations in cloud coverage. (Flight scheduling was such that only one flight was made successfully during a very light rain. This flight reached an altitude of 39,751 feet. Another flight launched in heavy rain was forced down by a sudden squall shortly after launching. By the time a second flight was prepared, the rain had ceased, thus it was not determined whether the balloon was forced down by the momentum of the rain or because of an error in making up the free lift.)

E. ASCENSIONAL RATE

1.) Theoretical Considerations

When a balloon is released from rest (velocity = 0) the forces acting on it are its weight and the buoyant force of the air. Since balloons are filled with a lighter-than-air gas, in this case hydrogen, the resultant force is an upward one and is expressed by:

$$F = V (Da - Dh) - (w + W)$$
 (1)

Where: F = Free lift (grams) or upward force.

V = volume of balloon (c.c.)

Da = density of air (grams/c.c.)

Dh = density of hydrogen (grams/c.c.)

w = weight of balloon (grams)

W = payload (grams)

If F>0 the balloon is accelerated upward and as a result of this acceleration it acquires an upward velocity and therefore experiences a retarding force.

The magnitude of the retarding force is dependent on the properties of the fluid (density, viscosity) and on the velocity, shape, and size of the body passing through the fluid. Through an analysis beyond the scope of this report the law of motion of a body in a turbulent fluid is given by*:

$$F = (k/g) Da v^2 A$$
 (2)

Where: F is the retarding force in grams; k a dimensionless constant depending on the shape of the body, in this case a sphere, and on the Reynold's number; g the acceleration of gravity; Da the density of the air in grams/c.c.; v the velocity in cm/sec; and A the cross-sectional area in cm² of the body

^{*} Clarke and Korff (Journal of the Franklin Institute "The Radiosonde" Oct. 1941)

normal to the direction of flow.

As the velocity increases, the retarding force also increases and eventually a velocity is reached such that the upward force and the retarding force are equal. The balloon then ceases to accelerate and moves with a constant velocity called its terminal velocity. This velocity can be found by setting the upward force equal to the retarding force.

Since $V = 4/3 \pi r^3$ and $A = \pi r^2$ where r is the radius of the balloon in cm. we can rewrite equations (1) and (2) thus:

$$F = 4/3 \pi r^3 \text{ (Da-Dh)} - (w + W) \text{ (eq. 1)}$$

$$F = (k/g) Da v^2 \pi r^2$$
 (eq. 2)

Combining them we get:

$$\frac{F + w + W}{F} = \frac{4 \pi r^3 (Da-Dh)}{3 \pi r^2 Da v^2 k}$$
or: $v^2 = 4/3 g \left(\frac{Da-Dh}{Da}\right) \left(\frac{F}{F+w+W}\right) \frac{r}{k}$ (eq. 3)

From equation (1) we see that

$$V = \frac{F + w + W}{Da - Dh} = 4/3 \pi r^3 \text{ thus } r = (3/4\pi \frac{F + w + W}{Da - Dh})$$
(eq. 4)

Replacing r in eq. 3 by eq. 4 we get:

$$v^2 = 4/3 \text{ g } \left(\frac{\text{De-Dh}}{\text{Da}}\right) \qquad \left(\frac{\text{F}}{\text{F+W+W}}\right) \qquad \left(\frac{3/4\pi \frac{\text{F+W+W}}{\text{Da-Dh}}}{\text{k}}\right)^{1/3} \qquad (\text{eq. 5})$$

Solving equation (5) for v we get

$$v = \frac{85}{(k)^{1/2}}$$
 $\frac{(F)^{1/2}}{(Fw+W)^{1/3}}$ cm./sec. at sea level (eq. 6)

Values of the resistance coefficient, k, depend on the shape of the

body, in this case a sphere, and on the Reynold's number, $R = \frac{Vr}{U}$, where v is the velocity in cm/sec.; r is the radius in cm.; u is the kinematic viscosity (viscosity/density) of the fluid, air in this case.

It is evident that:

- (a) according to equation (3), the initial rate of rise depends largely on the values of r and k for each balloon.
- (b) the properties of the atmosphere are continuously changing with each increase in altitude (density, viscosity and temperature decrease with altitude).
- (c) The properties of the atmosphere are variable at different locations, time of day and season of the year. (Temperature, convection currents, winds, solar radiation etc. are variable.)
- (d) The properties of the balloon are continuously changing with altitude (radius increases with altitude, membrane tends to stiffen with decreasing temperature, deformation of shape during ascent, etc.)

Thus it can be seen that the ascensional rate will not necessarily remain constant throughout the flight nor necessarily be the same for a given balloon at all times and places.

Since R, the Reynold's number, depends largely on the value of density of the air, it can be seen that the value of R will decrease continuously with each increase in altitude. Examination of figure 5 (k versus Reynold's number) reveals that, for balloons starting off with a Reynold's number of less than about 1.2×10^5 , k will remain constant throughout the flight and according to equation (3) velocity will be proportional to the square root of the radius, hence will increase with increasing altitude. In those cases where R is greater

than about 1.5×10^5 , k will not remain constant but at some point during the flight will increase abruptly when R becomes less than about 1.2×10^5 . Balloons of this type will have a lower rate of rise in the upper portion of their flight than in the lower portion. If R is very much greater than 4×10^5 then it can be seen that the abrupt change in k may not occur before the flight terminates.

It has been reported by some investigators (8) who have studied ascensional rates of various sizes and types of meteorological balloons that the ascensional rate.

- (1) is greater during the day than during the night. (We believe that this phenomenon is due to the effect of solar radiation on the temperature of the gas inside the balloon and the tension of the balloon film during flight. Since the gas is warmer at any given altitude during the day than during the night, the net result is a larger volume during the day and hence from equation (1) free lift is greater during the daytime at any given altitude.)
- (2) is greater in the afternoon than in the morning and is affected by local topography. Presumably, these phenomena are due to the variations, with time and place, of convection currents and temperature.
 - (3) is affected by changes in wind speed and direction.
 - (4) is related to the material and shape of the balloon.
 - (5) is related to the kind of gas used in filling the balloon.

In view of all these factors, it is evident that any formula based purely on the physical laws of motion to predict the rate of rise of an expansible balloon can at best be only a close approximation since it is impossible, within the scope of this report, to take into account all of the variables included

in the motion of a dynamic body in a dynamic fluid.

However, for practical purposes, we may utilize the data obtained from the flight tests in Task II of this report and arrive at a general formula for the Darex J-100 balloons which will be accurate in most cases to within 10% of the average actual rate.

II. Experimental Data:

(a) Average rate or rise for the J-100 balloons in the range under consideration; the empirically derived formula for rate of rise is:

340
$$\frac{(F)^{1/2}}{(F+w+W)^{1/3}}$$
 ft. per minute = average rate of rise

For J9-10-300 balloons carrying payloads in the range of 5 lbs., the empirically derived formula is:

420
$$\frac{(F)^{1/2}}{(F+w+W)^{1/3}}$$
 ft. per minute = average rate of rise

For J9-10-300 balloons carrying payloads in the range of 10 lbs. the empirically derived formula is:

500
$$\frac{(F)^{1/2}}{(F+u+W)^{1/3}}$$
 ft. per minute = average rate of rise

A comparison between the ascent rates predicted by these formulae and the actual ascent rates observed in flight tests indicates fairly close agreement, except of course in the case of night flights where, as has been previously mentioned, the observed ascent rates were slower than theoretical. It can also be seen from Table II. that early morning flights tend to be slower than afternoon flights in most cases.

(b) Incremental rate of rise

A study of the incremental ascensional rate characteristics recorded in Chart A. reveals the following:

- (1) In the case of the J9-10-300 balloons
- (a) In general the rate of rise is greater in the lower portion of the flight than in the upper portion. This is thought by us to be due to the relationship between Reynold's number and the coefficient of friction. That is, we believe the coefficient of friction increased abruptly at a point during flight when the Reynold's number decreased below about 1.2×10^5 .
- (2) In the case of the J-100 balloons
- (a) With the lower free lifts, the ascent rate tends to increase with altitude. We believe this to be due to the fact that these balloons start off with a low Reynold's number thus k remains constant, hence velocity becomes proportional to radius. Another factor may be super-heating of the gas due to excess time in flight.
- (b) In the case of the cloudy weather flights the rate of rise appears to be initially slightly slower than the latter part of the flight. It is believed that this is due to the lack of direct solar radiation received by the balloon from ground level until it breaks through the cloud layer.
- (c) Ascent rate during night flights is slightly higher in the initial part of the flight than in the latter part. We believe

this is due to the tendency of the balloon film to stiffen in the low temperatures of the upper portion of the flight, as well as the fact that the internal temperature is slightly lower than ambient, which in our opinion brings about a slight decrease in free lift.

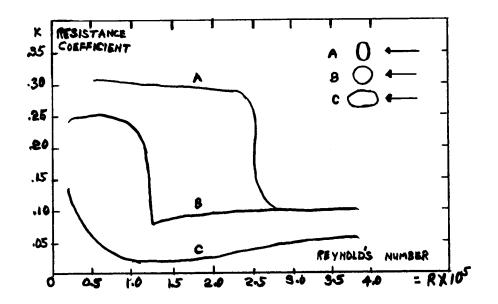


Fig. 5. Variation of resistance coefficient k with Reynold's Number R vr/v, for a sphere (B) and for ellipsoids of revolution with their major axes parallel (C) and perpendicular (A) to the direction of motion through the air.

TABLE 10.) RELATEONSHIPS USED IN CALCULATING RATE OF RISE FORMULA

Density of air at 20 °C = 1.025 grams/liter

Density of hydrogen = .09 grams/liter

Gravitational constant = 980 cm./sec.

Viscosity of air _ .0181 10 grams/cm.sec.

Kinematic viscosity of air = viscosity/density = .151 cm. /sec.

Reynold's number veloc. of balloon(cm. per sec.) rad. of balloon (cm.)

Kinematic viscosity (cm.2/ sec.)

Figure 5. Graph Reynold's number vs coefficient of friction.

TABLE II.) THEORETICAL VERSUS OBSERVED ASCENSIONAL RATES

J9-10-300 BALLOON

TIME OF DAY	GRAMS FREE LIFT	GRAMS GROSS <u>LIFT</u>	THEORETICAL FT./MIN.	OBSERVED FT./MIN.	PAYL	<u>OAD</u>
1:30 P.M.	300	2920	510	584	-	lbs.
9:00 A.M.	300	2920	510	509		11 tr:
6:50 A.M.	300	2920	510	433		11: 11
1:10 P.M.	300	2920	510	<u>526</u>		VI
			Avera	age 513		
11:30 A.M.	300	5190	500	483	10	lbs.
1:30 P.M.	300	5190	500	478		Ħ
5:26 P.M.	300	5190	500	519		11
11:00 A.M.	300	5190	500	<u>548</u>		n
			Aver	age 507		
12:21 P.M.	700	5590	745	722	10	lbs.
8:30 A.M.	700	5590	745	5 96		Ħ
10:00 A.M.	700	5590	745	697		11
11:40 A.M.	700	5590	745	704		n
			Aver	age 680		
3:32 P.M.	1000	3620	865	815	5	lbs.
6:00 P.M.	1000	3620	865	827		11
8:30 A.M.	1000	3620	865	744		11
11:40 A.M.	1000	3620	865	<u>834</u>		n
			Aver	age 805		
11:50 A.M.	1000	5885	875	879	10	lbs.
1:45 P.M.	1000	5885	875	924		Ħ
4:15 P.M.	1000	5885	875	914		Ħ
5:55 A.M.	1000	5885	875	<u>737</u>		12
			Aver	age 862		

J-100 BALLOONS

Flight Conditions Day Clear	Time of Day	Grams Free <u>Lift</u>	Grams Gross Lift	Theoretical Ft./Min.	Observed Ft./Min.	Pay- Load
W W	8:30 A.M. 9:30 A.M.	50 50	1080 1080 1080	235 235 235	251 252 276	2 Lbs.
" "	9:12 A.M. 9:30 A.M.	50 50	1080	235	<u>257</u>	Ħ
				AV	260	·
Ħ	8:45 A.M.	50	1535	207	187	3 Lbs.
n	12:40 P.M.	50 50	1535	207	273 220	# #
n n	8:30 A.M. 8:15 A.M.	50 50	1535 1535	207 207	227	11
 N	8:42 A.M.	50 50	1535	207	228	**
	0142		~~~		227	
				Ave	3. 232	
n	10:40 A.M.	100	1130	325	337	2 Lbs.
tt	1:05 P.M.	100	1130	325	342	11
n	3:40 P.M.	100	1130	325	322	11
n	11:35 P.M.	100	1130	325	336	#
n	2:45 P.M.	100	1130	325	<u>373</u>	Ħ
				Ava	342	
11	8:30 A.M.	100	1585	290	312	3 Lbs.
*	8:25 A.M.	100	1585	290	324	**
n	9:00 A.M.	100	1585	290	308	11
#	10:45 A.M.	100	1585	290	326	10
Ħ	1:15 P.M.	100	1585	290	<u>317</u>	11
				Ave	317	
11	11:00 A.M.	200	1230	446	434	2 Lbs.
17	8:30 A.M.	200	1230	446	417	11
Ħ	11:25 A.M.	200	1230	446	452	11
n	3:30 P.M.	200	1230	446	453	n n
n	1:30 P.M.	200	1230	446	449	
				Ave	g. 441	
n	11:00 A.M.	200	1685	405	423	3 Lbs.
Ħ	2:00 P.M.	200	1685	405	405	tt
Ħ	3:55 P.M.	200	1685	405	39 9	11
11	8:55 A.M.	200	1685	405	374	11 19
#	10:55 A.M.	200	1685	405	<u>405</u>	**
				Av	g. 401	

J-100's (Continued)

Flight Conditions Day Clear	Time of Day	Grams Free Lift	Grams Gross Lift	Theoretical Ft./Min.	Observed	Pay- <u>Load</u>
11	11:50 A.M.	200	2138	372	433	4 Lbs.
11	10:45 A.M.	200	2138	372	429	lt .
35	1:50 P.M.	200	2138	372	363	**
••	3:55 P.M.	200	2138	372	408	14
n	8:40 A.M.	200	2138	372	<u> 363</u>	#
				Ave	400	
*	1:45 P.M.	200	3047	315	344	6 Lbs.
n	3:30 P.M.	200	3047	315	<u>382</u>	-
	J. JO 1 1 14	200	JU41	727	2	
				r A .	rg• 363	
•	1:00 P.M.	320	3047	420	477	5.7 Lbs.
Ħ	10:40 A.M.	320	3047	420	465	**
•	2:00 P.M.	320	3047	420	<u>480</u>	71
				Āvē	g• 474	
11	2:15 P.M.	350	1380	570	558	2 Lbs.
 n	4:00 P.M.	3 5 0	1380	570	553	#
18	3:45 P.M.	350	1380	570	525	11
*	11:45 A.M.	3 5 0	1380	57 0	556	tt
*	2:15 P.M.	3 5 0	1380	570	<u>561</u>	29
				Ave		
•	12:05 P.M.	350	1835	520	545 53.0	3 Lbs.
•	11:00 A.M.	350	1835	520	512	11 11
•	9:00 A.M.	350	1835	520	510	ur
Ħ	10:00 A.M.	350	1835	520	<u>542</u>	
				Av	g• 527	
Ħ	10:30 A.M.	500	1530	660	675	2 Lbs.
Ħ	1:00 P.M.	500	1530	660	649	THE STATE OF THE S
n	3:05 P.M.	500	1530	660	627	W?
16	5:00 P.M.	500	1530	660	611	16
Ħ	10:30 P.M.	500	1530	660	655	39
		, , ,	-,,,			
				Av	g• 643	

J-100's (Continued)

Flight Conditions	Time of	Grams Free	Grams Gross	Theoretical	Observed Ft./Min.	Pay-
Day Clear	Day	Lift	Lift	Ft./Min.	FU. FAIR	Load
**	3:00 P.M.	500	1985	600	680	3 Lbs.
11	2:20 P.M.	500	1985	600	650	11
#	10:15 A.M.	500	1985	600	660	Ħ
Ħ	10:15 A.M.	500	1985	600	598	11
11	3:32 P.M.	500	1985	600	<u>585</u>	11
				Avg.	635	
n	8:34 A.M.	500	2440	565	580	4 Lbs.
Ħ	8:50 A.M.	500	2440	565	530	124
11	8:50 A.M.	500	2440	565	535	11
				A v g•	548	
n·	10:00 A.M.	620	3350	570	6 5 6	5.75 Lbs.
#	3:04 P.M.	620	3350	57 0	623	11
Ħ	1:05 P.M.	620	3350	570	674	it
				AVg.	651	

J-100's (Continued)

Flight Conditions	Time of	Grams Free	Grams Gross	Theoretical	Observ ed	Pay-
Night Clear	Day	Lift	Lift.	Ft./Min.	Ft. Min.	load
H	11:00 P.M.	200	1685	405	299	3 Lbs.
n	9:00 P.M.	200	1685	405	315	W
Ħ	9:30 P.M.	200	1685	405	346	*
19	11:45 P.M.	200	1685	405	347	11
Ħ	2:03 A.M.	200	1685	405	291	181
11	4:25 A.M.	200	1685	405	308	rt .
10	1:00 A.M.	200	1685	405	340	27
n	3:20 A.M.	200	1685	405	338	77
11	10:40 P.M.	200	1685	405	344	No.
Ħ	1:00 A.M.	200	1685	405	<u> 325</u>	**
				Avg	. 325	
Night Overcast	9:30 P.M.	200	1685	405	308	3 Lbs.
11	9:15 P.M.	200	1685	405	383	270
Ħ	9:00 P.M.	200	1685	405	357	90
				Avg	. 349	
Heavy					4.00	0.71
Day Clouds	11:50 A.M.	200	1685	405	420	3 Lbs.
11	3:30 P.M.	200	1685	405	416	11
11 11	1:30 P.M.	200	1685	405	446	M
	11:30 A.M.	200	1685	405	435	*
n	2:00 P.M.	200	1685	405	427	**
Light				Avg	. 429	
Day Clouds	3:20 P.M.	200	1685	405	425	3 Lbs.
u .	9:00 A.M.	200	1685	405	400	n
				Avg	. 412	

F.) ROOM TEMPERATURE DIFFUSION RATES OF J-100 BALLOONS INFLATED TO FLIGHT DIMENSIONS

The permeation of gases through membranes has been the subject of considerable experimental investigation. This process of permeation of a gas through a film is generally regarded as a combination in series of; a) adsorption and solution by the gas into the membrane at one surface; b) diffusion of the gas through the body of the membrane and; c) dissolution and desorption of the diffusing gas out of the membrane at the other surface. (5). The rate at which a volume of gas will diffuse through a membrane is dependent on temperature; pressure; area of membrane; and thickness of membrane. The rate of diffusion is proportional to the temperature, pressure difference and the area and inversely proportional to the thickness of the film. Because permeability is a combination of the two functions of solubility and microporosity of the film, it is evident that the rate of permeability will rise and fall with the temperature. An increase in temperature increases the vibrations of the moleculesmaking up the polymeric membrane. If this is regarded as a multi-layered reticulum, then it can be visualized that the interstices or holes are opened and closed more often or possibly to a greater extent to increase the possibility of the passage or diffusion of the molecules of gas through them. An increase in temperature also brings about an increase in the kinetic energy of the gas molecules allowing a greater frequency of collisions between the gas molecules and the walls of the film.

In the case of meteorological balloons it is well known that loss of lift due to permeation in flight is so small as to be negligible (4). This is understandable since as the balloon rises into regions of low temperature both the solubility and the kinetic energy of the gas are greatly decreased and the porosity of the film itself is markedly decreased due to stretching and orientation of the polymeric structure of the membrane. But, although it is generally conceded that permeation is a negligible factor during actual flight, it is operationally desirable to

know how much free lift would be lost to permeation, by a balloon inflated to flight dimensions and allowed to stand at room temperature for a period of time prior to launching. With a knowledge of the diffusion rate at room temperature of a balloon carrying a specified payload, and free lift to attain a desired ascensional rate, the free lift could be increased to compensate for future loss in cases where it is desirable or necessary to delay the launching of an inflated balloon. With this in view, an experiment was conducted whereby loss of lift due to permeation was measured, over a period of time, for J-100 balloons carrying payloads of 1, 2, 3, 4, and 6 pounds.

The procedure for the experiment was as follows:

Several J-100 balloons with 1, 2, 3, 4, and 6, 1b. loads attached were inflated with hydrogen to a point where the load was just balanced in midair. The balloons were checked at intervals of ten minutes and ballast was removed to restore the balance when lift was lost. The removed ballast was weight and recorded as grams of lift lost. This procedure was continued for several hours with the following results:

TABLE 12.) LOSS OF LIFT DUE TO PERMEABILITY J-100 WHITE BALLOONS

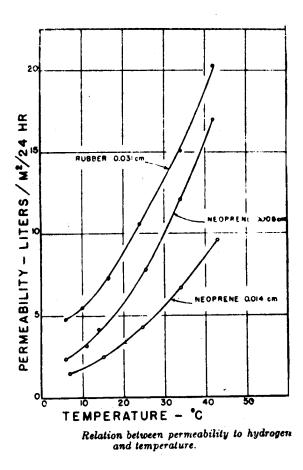
Payload	1 0	irams.	of li	.ft lo	st (1	O min	ute	inter	vals)	20-9	7			Avg. Loss (gms.) per hour
1 1b.	0.6	1.6	2.6	3:0	3.2	1.7	0:0	117	2:0	2:2	2.2	2.2	2:0	11.4
2 lb.	0.9	3.7	2.3	3.3	2.1	2.7	2.6	3.3	2.6	6.3	6.3	2.5	2.7	17.5
	1.9	3.1	4.6	4.6	3.8	4.8	5.0	5.3	4.4	4.4	4.4	5.3	5.3	25. 8
4 1b.	0.0	5.9	5.7	5.7	7.4	4.6	5.9	5.9	7:2	7.2	5.1	6.8	7.0	33. 0
6 lb.	0.0	0.0	7.3	7.3	9.2	9.9	8.8	12.3	8.5	8.5	10.2	11.4	10.2	48.4

It can be seen from Table 12, that the loss of lift in the first ten minutes is very small regardless of the load the balloon is carrying. We feel that this effect is due to two factors.

- 1.) During inflation the gas is cooled by adiabatic expansion and takes time to reach equilibrium with room temperature.
 - 2.) There is a short build-up period during which the film is not

saturated with the dissolving gas. Once the proper concentration gradient is built up, the gas will diffuse through the film at a fairly constant rate.

The discrepancy in the amount of lift lost between 10 minute intervals is probably due to minor disturbances in the air caused by slight updrafts or downdrafts in the area where the experiment was conducted. However, since these conditions are similar to those encountered in the field, it is believed that the average loss of lift as determined by the experiment is representative of the actual state of affairs regarding the practical relationship between diffusion and free-lift.



For the sake of convenience in operational use, a graph has been prepared wherein loss of lift in grams per hour has been plotted as a function of gross lift. (See figure 5.)

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Fig.



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G.) ADIABATIC COOLING OF INFLATION GAS

Two attempts were made to measure the temperature of the inflation gas versus room temperature immediately after inflation. The balloons used in the experiment were inflated at a regulator pressure of 25-50 p.s.i.

Attempt #1. A thermometer was inserted through the valve of the neck-closure into the balloon immediately after inflation. It was observed that the temperature inside the balloon was 1°C. below that of room temperature.

Attempt #2. Since it was felt that the thermometer used in Attempt #1, was not sensitive enough to accurately measure the gas temperature, a second attempt was made using a radiosonde. This was accomplished by attaching a long wire to the thermistor and inserting it into the balloon before inflating. Thus the temperature of the gas was measured while the balloon was being inflated. Again a difference of only 1°C, was noted.

From these experiments it was concluded that variation in lift due to adiabatic cooling of the gas was not a large factor in determining free-lift in the case of a J-100 balloon.

H.) CONCLUSION

It is concluded from the foregoing ground and flight test data that the characteristics of the J-100 and J9-10-300 gram balloons are:

1. Burst Diameter:

- a.) Ground burst diameters are somewhat larger than flight test burst diameters because of the effects of time and temperatures. (96 inches vs. 77-85 inches).
- b.) Balloons having a high ascensional rate attain larger burst diameters than balloons having a low ascensional rate.
 (77 inches vs. 85 inches).
- c.) Balloons carrying a payload of sufficient magnitude to preclude their ascending to high altitudes burst at a larger

diameter than balloons capable of ascending to high altitudes because of the difference in temperatures.

d.) Balloons can be stored for a period of 12 to 15 months without the necessity of pre-heating.

2. Burst Altitude:

- a.) In accordance with the gas laws, balloons with light payloads ascend to higher altitudes than those with heavy payloads.

 However, because of the relationship between time, temperature and burst diameter, balloons with a given payload and low free-lift will reach approximately the same altitude as a similar balloon with the same payload and high free-lift. The altitudes attained by the balloons under consideration range from:
 - (1) 23,000 feet with a 6 lb. payload to 40,000 feet with a 2 lb. payload using J-100 balloons.
 - (2) 45,000 feet with a 10 lb. payload to 58,000 feet with a 5 lb. payload using J9-10-300 balloons.
- b.) The presence of clouds does not affect the burst altitudes attained by J-100 balloons.
- c.) J-100 balloons flown at night reach the same altitude as those flown during the day.

3. Ascensional Rate:

a.) It has been shown that the ascensional rate of meteorological balloons is proportional to (Free Lift) 1/2

(Gross Lift) 1/3

and that for a J-100 balloon the theoretical rate of rise is equal to 340 (Free Lift) 1/2 feet per minute and for a (Gross Lift) 1/3

J9-10-300 balloon the theoretical rate of rise is equal to

(Free Lift) 1/2 feet per minute.

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- b.) Because of certain uncontrollable variables in the nature of the fluid and the body in motion, it was found that the above formulae are only accurate to within 10% of all the actual rate of rise, therefore
- c.) reliance must be based on experimental data if greater accuracy is desired.
- d.) Night flights rise more slowly than day flights.
- e.) Early morning flights rise more slowly than afternoon. flights.
- f.) J-100 balloons flown under conditions of overcast skies have the same average ascensional rate as those flown in clear weather.
- g.) Aged balloons have the same ascent rate as fresh balloons.

4. Permeability:

- a.) Permeability of a gas through a neoprene membrane is a solubility phenomenon.
- b.) The rate of permeability is directly proportional to temperature, surface area and pressure difference and inversely proportional to the thickness of the film.
- c.) Because of the rapid decrease in temperature during flight, the loss of lift due to diffusion during flight is very small, becoming negligible at high altitudes.
- d.) Because of a lag period, balloons inflated to flight dimensions and held at ground level will not lose any lift due to diffusion in the first ten minutes. However,
- e.) if balloons are inflated and held at room temperature for an extended period of time before launching they will lose from

11.4 grams of lift per hour for a 2 lb. load to 48 grams of lift per hour for a 6 lb. load. The rate of diffusion increases as the load is increased because the film is thinner with the heavier loads.

- 5. Adiabatic cooling of inflation gas
 - a.) If balloons are inflated at a regulator pressure of 25-50 p.s.i. there will be no loss of lift due to adiabatic cooling of inflation gas.
- 6. Uniformity of performance

The data obtained from the flight test program indicates that the average range in burst altitude for the J-100 balloons is 5,582 feet and the average standard deviation is 2,610 feet.

For the J9-10-300 Balloons, the average range is 7,400 feet and the average standard deviation is 3,610 feet.

DISPERSION ANALYSIS

J-100 Balloons

Payload (Lbs.)	Free Lift (Grams)	No.Flights	Range	Standard <u>Deviation</u>
2	50	4	5036	2450
2	100	5	9320	4050
2	200	5	7090	3020
2	350	5	3652	1570
2	500	5	3631	1560
3	5 0	5	8366	3600
3	100	5	7369	3170
3	200	5	8622	373.0
3	350	4	6772	3300
3	500	5	8385	3650
4	200	5	10905	4700
4	500	3	4877	2890
6	200	2	2733	2420
5.75	620	3	3314	1960
3 light clouds	200	2	400	3 5 8
3 heavy clouds	200	5	4786	2060
3 night flights	200	10	5850	1900
3 cloudy night	200	3	761	450
3 preheated	200	8	<u>8238</u>	2890
		Average	5582	2610

Average Range = 5,582 feet

Average Standard Deviation = 2,610 feet

DISPERSION ANALYSIS

J9-10-300 Balloons

Payload (Lbs.)	Free Lift (Grams)	No.Flights	Range	Standard <u>Deviation</u>
5	300	4	4033	1960
5	1000	4	12979	6300
10	300	4	10020	4860
10	700	4	2352	1140
10	1000	4	7736	<u>3750</u>
		Average	7400	3610

Average Range = 7,400 feet

Average Standard Deviation = 3,610 feet

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<u>.</u>	15,000 20,000 20,000 25,000 25,000 30,000	18.5 11.6 17.6	11.E 11.7	11: E 11: E 11: E	11.5 13.3 11.5	10.5	11.5 11.7 11.1	9.7 9.7 8.3	5.7 5.4 11.1	5.4 14.5 50	5.3 5.6 8.6	9.2 9.6 9.6	9.5 9.6 9.8
	35,000 35,000 35,000 26,000 26,000 25,000	11.3 E.7	18.8	//// //// Sanitize	5.5 11.6 d Copy Approv	9, 7. ed for Release 2011/0))/())5/03 : CIA-RDP78-03	€. € €. 7 8639A0005000	€.7 €.€ 040001-5	8,5 9,2	E (. E.E S.7	7.6 E.6	8.4 6.7

 1			i		1 11	Averege of 5 Dispite			1	1
Assertion No.	#1//	150		#52	11.5	10 2 1416 1162				
			7	8	2					
Faylout (1bs.)	7	9		600	600					
Free lift (grand)	500	1530	600	1530	1880					
(mass 1964 (grand)	16.20	1/2/6)	13.00	1.5						
	7-12-87	5-16-57	7-16-57	7-10-87	7-22-57					
Date	11130217	1:00 675	SEE THE	Steefes	18:30 F/2					
Stert	11: 25/13	2:03/11	1.88 613	GUI FIN	187 612				1	
Finish	\$7.8	63.7	66	61	67	61				
line in Flight (minutes)										
 52 / 51 (GI)	211/207	180/313	de/30	271/200	273.4/201				1	
F ₁ / F ₂ (additions)	155/200	100/155	101/501 101/200	us/ Ess	1111/225					
famet Altitude (ft.)	35,000	49.862	1,0,068		\$7,503	38,018			1	
eye, here of Riss (fil/s)	1683	6.65	687	611	658	613				1
										1
Impat Dieneter (de.)	63	E. 6	84.5	60.6	81.5	87.8		e la		1
Iniret Volume (entite feet	175	166	18:1	161	166	17%				
					1 1					
 Lete of Fire (ft./him.)			۔۔ را		E CF	153	-1			1
550 to 5,000	1010	704	622	616	617	668				
5,000 10,000	605	581	685	330	Sec	606		4.1		
10,000 15,000	784	603	661	574	CES	ØE.				
5,000 20,000	637	617	888	562	616	566				
25,000 25,000	677	689	685	616	715	647				
5,000 30,000 0,000 35,000	686	686	KSF	616	675	635				
	62.8	655	616	688	648	633				
5,000 A0,000		6.28								
								<u> </u>		1
Minutes in Flight										1
550 to 5,000	410	6.8	6.6	63	6.6	5.8				
5,000 10,000	3.0	6.5	8.0	8.8	8:1	7-7				
10,000 15,000	631	6.6	8.8	8.0	6.5	8.3				
20,000	68	8.3	7/6	8.7	6.0	8-7				
a,000 25,000	9,6	8.6	5.1	8.8	8.8	6.6			1	
30,000 30,000	11.17	7)3	1.0	6.8	7.1	7.7				
0,000 35,000	8,0 7.1	7.2	8.4	8.7	7.6 5.5	7.5				
35,000 10,000	- ""	7.8	8.8	₹.€	""	6.3				
45,000		1.6		1	l l					1
	1700									

<u></u>	Assembles No.	#3F:	#37	113.	115	112		Iverage 5 flaguage	11	129	<i>[3</i>)	10	13		Iverage S filghts
	Front (1to.) Front 14th (grand) Group 14th (grand)	3 51 18 85	3 50 1686	3 60 1888	3 50 1515	168E 60 3			3 100 18 EE	3 100 1588	3 100 1888	3 101 160 E	3 165 168		
	lute Stert Plates. Flue in Flight(ft./hdus	11:25/11	160	631 AM 1044 AM 144	11:81 AR	14.62 VW 11.62 VW 14.6		126.8	1030 AFT 1030 AFT 103-7	6-88-57 828 AB 1833 AB 184	1166 11:67 AR	10.45 KB 1897 GB 107	7-14-87 1118 F14 2348 F14 114		ne.
	T ₂ / T ₁ (GI) F ₁ / F ₂ (additions)		1118/515	1015/250 1015/250	18.88	115/253			11/288	हरत्त्र शृहहड हरत् ।इ।	nalege nalege	100 /251 d 100 /251 d 1259	ustres	AND PERSONAL PROPERTY AND PARTY.	
<u> </u>	Enral Attitude (ft.) Eve. Lete of Elec(ft/att	163	38,839	21, 778	28, 200	33,163 381		33,666 FF7	24,173 218	386	35, 686 301	34,678	36,005 317		36,488 215
	Suret Diemeter (in.) Suret Volume (cutic feet	78.6 188	87.4 171	75 E 138	75.0 161	76.6 188		99.4 143	54 150	€7 70₹	8% 188	78. E 163	160 160		(11.3 161
	Sect. of Rise (43./dm.) Sign	165 166 167 132 231	259 265 265 265 265 275	188 218 187 287 288	184 203 218 221 221	198 197 218 721 781		187 201 206 220 220	877 758 765 765 786	671 765 361 365 816	864 864 863 813	368 368 381 387	303 275 280 312 305		289 260 288 200
	5,000 30,000 30,000 33,000 5,000 46,000 6,000 45,000	\$ 84	373 333 336	298 38i	7.48 7.47	58°L 568		788 718	335 335	575 357 373 565	3.5 g 3.76 3.60	37/	221 221 221	***************************************	156 136 206 208
	Nimtes in F14ght SF0 to 5,000 5,000 10,000 0,000 15,000 5,000 20,000	36.5 36.5 36.6 36.0	16.8 28.7 20.9 20.9	30.7 78.0 78.0 28.0	76.6 76.6 73.7 75.6	28.0 287 255 545	- A-40	787 768 760 783	17.5 16.5 16.7 17.3	17.7 16.6 16.6 16.7	11.6 18:5 17:7 16:6	14.8 17.8 16.7 16.8	15.6 15.8 15.8 16.6		156
)	0,000 25,000 3,000 30,000 0,000 35,000 5,000 40,000 16,000 45,000	\$ & . (. 15. 5	16.6 16.6 5.7	61.E 16.6 E.G	20.6 20.6 12.6	15.5		86.5 17.7 11.7	16.8 16.1 18.8	165 187 184 29	15.1 188 14.7 7.3	14.7 14.7 18.8	16.8 13.5 18.5 3.6		163 183 140 188 41

5	Assension No.	#26	#1/.	11/4	F330	1,11	Average of 5 flitnite	115	#21	123	13/	Average of A fidge
	Paylord (No.) Pres 1881 (grand) Greek 1881 (grand)	8 200 1085	300 300 300	3 200 1015	8 200 1085	3 200 1065		3 356 1835	350 1835	3 250 1636	3 850 1885	
	Date Start Finish Time in Flight (nimites	mee an 1848 fee	7:15-57 2:05 17% 3:55 17% 9155	7-18- 87 8188 191 838 1111 88	\$14-83 (#\$ AM H131 AM 96.6	96 165E NO 1841 (N.	E\$+3.	6-21-89 12:05/75 1:19 /7: 7/	18558015 17 775 750 100	6-25-85 G180 AIS H116 AIS 76	7194 67 10100 AU 17150 AM 5819	67.8
]	78 / 73 (64) 33 / 38 (61114ters)	585/345 585/345	mos/ans	us por	100/200	100/208		1108/202	115 / ESE	131/78E 1881/888	111 / 500 1111 / 500	
<u> </u>	Burg Attitude (ft.)	31,361) 483	28,661	35,456	26, 165 378	30,691 418	88.31E GEV	37,538 548	83, 817 818	28 287 570	\$1,687 \$48	35,786
) D	kurst Dieneter (In.) kurst Volum (entic fer	76.2) 1.66	89.7 186	85.0 158	83.8 176	75.8 75.6	81.6 107.6	86.5 197.5	81.8 168	802 8.63	80.0 187	83.9 181
<u>-</u>]	Eate of Rise (ft./fits.) 5% to 5,000 5,000 10,000 10,000 15,000 15,000 20,000	391 391 442 435	480 363 378 368 417	287 287 351 218 468	268 327 368 373	708 208 308 351 250	4:11 863 874 258 408	787 821 458 456 450	501 156 156 660 640	478 458 480 485 E10	70 E 121 515 516 516	610 479 479 502 516
]	25,000 25,000 25,000 30,000 35,000 35,000 35,000 46,000 46,000 45,000	186	408	396 230 384	120 111 166	418 417	2.18 2.48	E18 550 510	2.62 2.62	543 F17 F57	476 567 ,	est est
]	Minutes in Flight 500 to 5,000 5,000 10,000 10,000 15,000 15,000 20,000	12.8 12.8 11.3	11.4 12.8 14.7 18.8	11/2 12/2 12/1	13.6 14.7 16.3 18.6	11:8 13:8 15:6 18:8	11:7 13:4 13:4 12:7	6.4 8.6 16.8 11.0	9, E 16.1 11.1 10.6	16.6 16.8 16.1	6. E 10 G 9. 7 9. 0	7.8 10.5 10.6 10.6
l)	20,000 25,000 25,000 30,000 30,000 35,000 35,000 45,000	17:0 11:0 9:8	18-6 18-6 11-8 8-6	18.4 17.6 11.6 11.7	18.6 11.8 12.2 8.8	17.6 17.4 2.j	18.8 18.0 11.3 1.7	16.7 8.7 8.1 6.6	5.5 5.1 6.6	5.6 5/2 5/3 61	9.0 11.5 2-0	9.7 9.3 9.0 ;F

					1		T 1				7
	Lecensian Mt.		818	/ 31,	#16	133	Average of Sillights	 		<u> </u>	1
	Paylout (1to.)	3	3	.3	3	.3					
	Free Lift (grans)	500	500	500	600	600					1
	Great Life (great)	1885	1868	1568	1868	1568					
	Date	6-18-85	6-20-57	1121-87	6-84-87	7-6-67		1	1		1
	Stert		F:18. 674	10:13 67	wife IM	838 FM					1
	Pintel	8:87 600	331 CV	11:53 /11		1137 115					1
	Time in flight (minutes		83.3	1,4	63	65	54.8			1	
	13/11(0)	115/208	185/206	1357 /258	1313/200	111/500					1
	F ₁ /F ₂ (atlisters)	111/216	1008/203	1115/316	un /ser	1885					
	Eurst Altitude (fi.)	36,600	34,680	30055	31, 650	37,518	34,786	1		1	1
	eve. Lete of Electricity		650	660	691	£ 8:E	635				1
	host blooter (it.)	683	65.5	61.3	68.1	65.6	8.5.4				1
	Burat Volume (calde fee	ACCUSED TO SELECT ON SHARE	150	165	150	215	181		1		
	Lete of liter (fi./sit.) SEC to 5,000	920	દર્સ્ટ	760	613	760	775	 1		l	-
	5,000 10,000	806	616	615	(115)	577	623				
	16,000 15,000	588	686	617	60%	514	581				1
	15,000 20,000	333	686	657	506	582	600				1
	20,000 25,000	663	686	61.8	586	533	6.00			1	1
	52,000 30,000	668	6.83	635	663	260	67.5				1
	30,000 35,000	676	688	689	350	61.6	627				1
	35,000 75,000 75,000 75,000	681				632					1
	70,000 75,000				300					1	
	Minutes in Filest SW to 5,000	6. 7	ر بر	6.3							
	5,000 10,000	6.8	3.8	E-3	7.E E./	6.3	3.3				1
	10,000 15,000	8.6	8.6	E-1	8.5	9.7	7.5 8.6				1
	15,000 20,000	8.5	8.6	7.6	8.8	6.6	8.3			I	
	20,000 25,000	8.3	8.0	7.8	8.4	8.4	8.3	1			
	25,000 30,000	7.5	7.5	7.4	6.3	6.5	8.0				
	30,000 35,000	2.4	24	.8	7.7	6.2	7.8				
2.15	35,000 /0,000	5.7		FEMALES I		1.6		1	1	1	1

Legenton En.	/ 35	-1 15-	156	150-	1/5	ा ५ १३४१४४ व्यक्त	/13	1 107	1%	45/11/
hyled (Us.) Has left (gras) Gras left (gras)	4 201 2148	G 200 218£	5 801 8138	{ \$00 \$15£	\$126 \$00 G		9 500 8110	1; 600 71110	# 500 8440	
Date Stert Finish Time in Flight(minutes)	1:0510		790-87 1185 FM 810 FM 71	7-11-E1 3:EE 171 5:25 F16 50	Exectle:	25.6	6-21-57 6:84 M 9: 25 M EE	6-24-81 881 AN 8:01 AN	6.50 km	\$4.4
52 / 53 (⁶ 1) 53 / 52 (atliffere)	15/204 105/250	ws/est	11/208		urs/ret		1418/251	161 /255	781/808 10:0/816	
Favet Miliate (ft.) eve. Fate_of kies_(fi.fe	38,811 () 583	37,107 LFS	88,680 368	36,87 E 201	27,687 363	80,776 600	\$1,677 .516	87,000 Esc	36, 873 535	28,50 EG
kurst likunter (in.) kurst ledum (subis fer	85.0) 185	&\$.6 167	78.0 150	88.E 816	88.6 188	180	88.6 818	63.8 177	86.0 184	88.9 189
Ects of Rice (ft./sin.) SEC to 5,000 5,000 10,000 10,000 15,000 15,000 20,000 20,000 25,000	527 354 384 384 455	688 618 886 488	565 561 561 563	667 688 887 887	366 366 366 366	445 385 365 366 350	675 545 616 617	506 510 510	\$.67 \$76 \$10 \$71	55\$ 556 550
25,000 25,000 25,000 35,000 35,000 46,000 46,000 45,000	487 447 469	488 488 418	36E 211	347 353 470 457	361	3.87 2.03 1.64	(C3E	6 3 6 6 3 6 5 7 9	841 841
Himter in Fight SP to 5,000 5,000 10,000 10,000 15,000 15,000 20,000	\$11 17:5 18:1 18:3	11/3 18:J 18:O	18.6 18.6 18.8	8.6 11.6 14.6 18.8	18.0 18.E 18.7 16.6	16.7 18.8 18.7 18.4	7.1 9.1 8.5 9.9	8.6 166 8.8 8.8	6.6 9.5 9.6 9.6	6.6 6.5 6.3 6.3
20,000 25,000 25,000 30,000 30,000 35,000 35,000 40,000 46,000 45,000	11.8 11.8 4.8	11.6 11.0 1.8	14.6 14,	14.4 18.7 11.5 8.8	18.6	18.5 18.4 1.7	6.1 5.0 3.4	9.4 3.8	5.8 5.3 1.5	5.7 5.1

Legention Et.	186	130	Average of 2 flagues		1 23	I ric	Average of 3 Dispits	
Feyland (18s.) From 14ft (grand) Grand 14ft (grand)	6 200 30 <i>0</i> 5	6 866 8667		6.76 686 3386	5.78 686 3350	5.75 686 3356		
Date Stert Finish Time in filight(plantes)	7-8-87 1: AEIN 2: 80 IN 0.E	7-9-89 3:30 (9) 4:53 (1) 68-7	03.5	6-81-E7 1020/AR 1028/AR 33.8	7:00 60		&5·4	
F ₂ / F ₃ (^E 2) F ₁ / F ₂ (atll4ters)	est/acs less lines	78/303 1010/398		સ્લ્ક/ક્લ લ્લોન્ડર	रहर्गडार स्टाहीसा	105/209		
First Attitude (ft.) eve, kete of rise(ft/sir	380	25,00	28,741 275	21782 687	88,166 675	28108	83,000 681	
Burst Diemeter (in.) Burst Volume (cathe fer	654) 183	810 810	60.E	88.0 810	6/.6 233	88.4 818	69.3 716	
 Rete of Rise (ft./site.) SE	337 378 388 388 331	488 463 316 378 383	213 401 317 370 3.57	\$76 617 567 566 650	775 985 510 599	683 668 633 536 516	758 778 567 585	
25,000 30,000 30,000 35,000 35,000 40,000 40,000 45,000								
Number in Fisch Site 5,000 5,000 10,000 10,000 15,000 25,000 20,000	10.2 10.1 15.7 13.6 7.8	8.6 11.6 15.6 13.8	11.6 18.3 15.6 14.5 11.5	6.9 6.7 5.8 8.5 7.7	6.5 9.9 9.9 9.9	7.6 7.6 7.8 8.7	6.6 6.5 8.5 8.5	
25,000 30,000 30,000 35,000 35,000 40,000 40,000 45,000								

		CHOOD COVER	Average of				land) anns		Average of
Assention by	26	1/3:	2.13573.46			135	116	#19:	S.Filtelitz
	2	3			3	3	5	3	
Pres lift (grans)	200	200		700	200	200	200	200	
(her lift (frem)	10.65	1086		1668	1005	1685	1686	1065	
In the	6-28-67	8.81.51	i		5-1:-57	9-11-57	10-7-87	10-5-59	
Stert	2:20 F 12:	9160 KK		ILT AT	33300		11:35 6.74	1:45710	
Finish.	3:45 110			11.28.61	4:58 FA		18: 50 CM	11.72 LA	
Time in Filght (viewtes	E5	es	E7	98.4	66.3	\$3,\$	60.5	SE: 60	90.3
15/11 (%)		eest fees			110 % Sec. 6	134/50	110/2015	221/267	
F ₁ /F ₂ (atllifiers)	1126/237	1115/836		millen	\$\$i/##3	les lies	101/885	rois (217)	
Emat Altitute	36,744	25,600	80,000	35)150	36,778	11,586	37 730	38,386	88, 786
svs. Lete of Else	1.808	355	 418	csi	416	ino	435	627	88, 788 489
First Dispeter (86.)	63.0	63.0	63.0	87.1	63.8	66.9	64.6	85.0	E5.5
Biret Volume (ouble fee		1716	1758	240	157	210	178	160	180
				111					
Lite of Rise (ft./him) SE to 5,000	355	38.6	370	88.7	1250	(411	661	683	989
5,000 10,000	210	375	355	386	357	1.17	368	385	335
10,000 15,000	1163	350	357	885	003	357	1:00	797	75/3
15,000 20,000	1.84	367	356	173	307	1.17	1.80	038	918
20,000 25,000	484	567	383	06!	9.67	1.84	1:27	413	1188
25,000 30,000	443	1.81	637	417	631	438	456	1.68	437
30,000 35,000	400	152	500	1.78	1.68	1.62	1.03	1969	4.57
35,000 40,000	418	573	587	\$18	465	156	515	688	4.75
70,000 72,000					<u> </u>	480			
Nimter in Fight	18.6	18.6	18.8		11.5	7.6	10.0	110	100
5,000 10,000	12.2	18.8	18.7	18.6	108	12.0	13.1	13.6	18.1
10,000 15,000	11.3	11.3	18.6	13.0	18.6	15.2	18.5	17.6	18.7
25,000 20,000	11.5	18.6	18.6	10.6	18.6	18.0	11.5	11.5	108
20,000 25,000	11.6	1/3/6	18.7	11.8	107	1116	11.7	18.1	11.5
25,600 30,600	11.3	11.6	11.4	18.0	11.0	11.6	10.8	11/3	11.5
30,000 33,000	10.4	616	11.1	11.0	11.5	106	10.8	11.8	10.8
35,000 40,000	8.5	11.6	118	8.8	8.8	11.6	5.4	7.4	7.6
 		! !	 <u> </u>		J	3.7.		11	

	1 .	183	133	Nata of	197	#85	Average of 2 fitehts	12	148	Average 2 flight
-	Assertion lit.	84 M.O.			84,0018	24 MIS.		18 17:11.	200018.	
	legical (165.)	.3	3		200	S00 8		200	3 700	
	Free last (grand)	701	1085		1015	1668		1015	1088	
	trans lift (grans)	NOEE _	West.		E'6 888			nent	MINT	
	Date	7-78-51	6-1-87		6-1-51	E-2-67		9-5-51	9-6-57	
	Stert	10:05 /11	8: St F20		15.61.132			10: 88 11:		
	Firsts	10:07 1.1			B. EF FF.	E: JE (M)		18:18/77		76
	fine in Flight (nimites	47	\$7.7	58.4	2.0	,,	75	2.5	E1/	18
	59 /51 (62)	101/855	182/301		1364/20	215/213		138/253	278/288	
	F ₁ / F ₂ (all libers)	1018 505	1018/128			1016/808		1115/383	un len	
	Burst Altitote	16,817	28,713	50,680	31,818	31, 657	31, 287	25, 274	112	31,30
	Leve. Bate of Rice	1,02	355		7.16	- Gra	416	-1-578	2.162	
	Envet Dienster (ie.)	66.6	20.1	65.8	75.0	78.0	775	768	80.0	78.1
١	Envet Volume (entite for		101	108	189	145	168	135	155	145
	Este of hiss (ft./str.) SFC to 5.000	355	360	328	658	349	400	608	657	434
	550 to 5,000 5,000 10,000	881	357	35 6	367	600	378	375	308	336
	10,000 15,000	391	378	338	375	387	366	350	3.67	368
	15,000 20,000	1.88	383	406	1.24	1.34	978	357	115	1.18
	20,000 23,000		410	1115	416	435	381	357	481	781
	25,000 30,000				135	1.28	11.81	1,60	451	138
	30,000 35,000				430	1.80	1.75		4.84	458
	35,000 46,000 46,000 45,000									
						L				
	Minutes in Flight								11.70	11.0
	5F0 to 5,000 5,000 10,000	18.6	12.4	18.8	16.6	11.5	16.7	14.3	12.0	18.4
	10,000 15,000	18.6	13.2	18.7	18.6	12.5	18.8	11.3	12.8	13.0
	15,000 20,000	6.3	12.7	18.8	11.8	165	11.7	18.6	18.3	17.5
	20,000 23,000		66	2.1	18.2	11.5	11.6	11.59	111.6	18.
)	25,000 30,000				11.6	11.7	11.6	9.8	111.6	11.1
	30,000 33,000				8.1	2.7	7.9		28	3.6
	35,000 70,000									
	10,000 15,000									
	160									

1	l.			İ		1108	4184	/119	erigis	
Ligario fo.	_ <i>!n</i> :	_! ii	18:	_#155:	14 1715	18 11:13	15 n.C	18 17:18		
Iga of Billion	120,02	13 1/18	13 17105	19 1718	8	3	3	3		
Tryled (16.)	3	.5		ios	200	200	800	201		
Free ISO (grans)	800	200	700				1086	10.00		
(Grank 1464 (Grank)	1065	10 8F	1086	1666	1065	1005	MARE	IN INT		
Ive modificating	NONE.	NUE	MING	rent	HUIT.	NING		8-18-87		
Inte	9-5-81	8.6581	9-6-57		9-28-51			830 110		
Steri	8:30 1.15	11:00 1:10	2:00 M	8200 KM		3:05 FM				
Finish	10:1.8/1	18:30 17	and Pa	W: 19 MM		C:53114		W: 20 VI	83.5	
Time in Flight (nimites	7.3	60	66	78	65	S.C.	81	56		
72 /13 (52)	84.5/ese	27/858	West	134/187	145/885	218/305	136/286	175/853		
F ₁ /i ₂ (stilliters)	2130	1008/207	1018/381	1015/253	un lest	millist	105/868	1108/837		
1,.,									85.56	
Invest Altitute (fi.)	27,128		59,816	32,017		60,000	88,018	36,756	38,60	
eve, hate of Riccitt/61) \$76	<i>C15</i>	351	GOS.	368	413	\$68	382	3.65	
	58.6	68.6	78.6	76.5	22.6	8778	78.6	83.8	58.3	
Durat Director (311,)		181	11/7	14.6	163	804	154	158	183	
Farret Velyme (earlie Fer) ire	/**	l "′	""	1 '5'					
Date of Rise (ft./air.)										
SF0 to 5,000	584	506	5.83	900	313	680	3.37	371	1,50	
5,000 10,000	811	368	367	380	381	367	347	388	\$150	
10,000 15,000	368	857	375	357	365	365	375	367	3).57	
15,000 20,000	328	411	368	385	865	357	308	376	371	
20,000 25,000	387	611	1110	285	367	413	366	351	350	
25,000 30,000	355	415		475	368	0.03	388	600	450	
30,000 35,000		1.75		633	875	486	1.35	410	410	
32,000 10,000						132		400		
10,000 15,000										
***************************************					1					
Minutes in Plight					1			18.5	10.0	
570 to 5,000	6.6	9,8	9,0	10.3	14.6	100	14.8	The state of the second state of the second	110	
5,000 10,000	16.6	11.6	12.8	18.6	18.6	186	14.4	14.8	16.8	
10,000 15,000	14.5	18.6	15.8	14.0	11.3	13.7	15.4	16.6	13.8	
15,000 20,000	13.8	18.8	13.6	180	18.7	18.6	18.6	13.3		1 1
20,000 25,000	18.8	18.5	13.5	13.1	15.6	18.11	13.6	18.8	18.8	
25,000 30,000	63	10.3		105	126	11/3	186	18.5	11.5	
30,000 33,000		8.3	ll i	4.6	3.6	1118	6.9	18.8	6.0	
35,000 40,000	l					18.1		8.6		
20,000 25,000	1	II .	li i			1				

	AYEE 1	II F2	ree 1	CLEAR MIAN	ena protes	Filterio E	<u>-</u>				Average of
		<i>1</i> 103	# 10!	#105	#10%	1100		010		#315	10, 1119,000
Accounter No. Legison (Ata) From Idli (grana) Gross Idli (grana)	3 70 1688	3 700 108C	3 210 106F	3 800 10 8 5	3 201 108€	10 EE 3 CO	3 201 1685	3 201 1085	3 FOI MUT	3 201 14 (5	
Dite Start Finish	7-15-67 11:0:09: 1:07 hm	5:00 TM 1:08 IN 108	&-6-87 \$130 FF 11:14 FF 104	11:45 0M 1:37 AM 107	119	4:48 1.11 6:17 1.11 118	8:61 M2 111	3:80 pm 5:08 pm 108	11.5	11.61 VA2 5: 25 VA2 11:61 VA2	1167
		238/25E 1018/207	105/855	283/291 1115/880	1111/255	1117/255	१६१/६६) १०११ /६६६	1011/251	१८४/१८४ १८४८/४८४	1011/263 1011/263	
Smat Altitude Eve. Lete of Flee Utilia	38,000			37, 113		39,461		50,866 538	365	30,411	388
harst blemster (ir.) Earst Volume (calde fe	8.50	78.0 148	EF. 8 150	175	10.8 81.0	16.5	E3.6 175	88.E 178	E11.6	158	176
Rete of files (ft./him.	,				1	175	400	367	386	360	3.63
550 to 5,000	131.6	383	387	355	875	203	387	236	331	305	310
5,000 10,000	276	315	3/1	370	318	318.	25%	315	325	378	355
10,000 15,000	889	355	315	367	315	158	305	315	336	3.05	3/3
15,000 25,000 25,000 25,000	284	279	3.5%	384	278	515	335	356	34.5	305	388
25,000 30,000 30,000 35,000 35,000 46,000 46,000 45,000	316 878 386	303	367 367 358	367 218 306	387	3.61	3,57	356	365	\$62	383
Nimites in Flight	1			18.0	16.1	16.6	11.5	18.5	18.4	13.8	3.51
SE0 to 5,000	11.14	15.6	15.3	11.6	17.9	16.5	15.5	14.5	15.1	16.4	15.8
5,000 10,000	18.6	16.8		15.6	160	16.6	17.0	16.8		16.6	15.4
10,000 15,000 15,000 20,000	15.6	15.4	14.8	13.5	16.4	16.5	10,5	16.5		16.6	16.6
20,000 25,000	17.0	183		18.9	17.5	16.6	148	15.3	11:5	16.6	15.3
25,000 30,000 30,000 35,000 35,000 40,000 40,000 45,000	16.4	76.3		10.0	15.7		14.8				\$.{ 18.8

	11 (S) ()	<u> </u>	e (100	n Gian Gian Mar 171	<u> </u>	 <u> </u>		II decrete v	•
Legyrtey Se.	120:	1281:	# 21%:	Arrigi 6	1	 	ļ	<u> </u>	1
Feyfood (No.) Free ISA (green Green ISA (green)	8 800 1066	5 800 108F	3 200 10EE						
 dete Sterf: Finish: Finish: flight (nimites	5131 194 11105 (15	SIFFI:	11 + 54-5 7 5; 60 F 15 6; 57 F 15 65	88					
7 ₂ / 7 ₃ (43) F ₃ / F ₂ (additions)	हरू भू _{र हर} १०५१ हरू	खदा हु। १००/इटर	ws/sec						
Egrat Altitude (fi.) art. Lete of Alex (fi/s	58,583 398_(1	\$6,6% \$6\$	\$600 \$87	\$6,018 368					1
Burst Dienster (in.) Burst Volume (cuids for Bets of Blex (N./hiv.)		E1.6 168	EJ. 3 16G	80.E 1FG					
570 to 5,000 5,000 10,000 10,000 15,000	306 868 333	316 356 357	366 366 36	\$64 3.48 \$5.6					
15,000 20,000 20,000 25,000	368	566	618 881	361					1
25,000 30,000 30,000 35,000 35,000 40,000 40,000 45,000	284 284	3 CF 364	418 881	336					
Fine in Fitzet (ntr.) SV to 5,000 5,000 10,000 10,000 15,000	17.5 17.5 16.6	11.5 18.6 11.7	12.6 12.7 16.7	18.1			,		
25,000 25,000 25,000 25,000	16.1	114. E 14. E 18. E	18.1 18.8	18.5		7			1
25,000 \$0,000 30,000 35,000 35,000 46,000 46,000 45,000	10.4	16.5	18:0	18.8					

Y					1	Average of	1		T ⁺	i i	Iverege of
4	Armedon Iv.	169		1101	100	A thinks				18	/ Filente
	Taylord (1to.)	.6	E	1	6		5	•		5	
	Free lift (grans)	300	\$00	300	200		1000	1666	1000	1000	
	Cross Idea (grams)	86.80	7580	8.980	8686		2080	3680	3680	3620	
	Dite	7.80.51	8. 8. 84		€-£-£1		5-11-83	3-28-57	7-25-57	7-30VE7	
	Sizei	11: 35077	- Internet Laurence manufacture and	GEE M	1:10 (1):		1:38 (1:	6:05 F/2	6:30 AV:	11122 111	
	Finis):	35.05.115		5:08 AM	8:63 ():		1330 F75	7:16 FM	8188 1.05	18:48 17	
	Time in Flight (intenter		108	157.6	113	111.5	66	7.6	65	60	70
	T2 / T1 (EX)	211/856		F18/866			F15/80E	11/256	F18/848	Enless.	
	F1 / F2 (militters)	nest les	melso	1011/28	101 /18		100/00	1018 16E	1015/115	nerifet	
- 19-1											
	Forst Militaic (feet)	56,888		58,616	ES, 616	66,167	54, 848	64,381	51,848	£1,788	53,608
	lverege hate of him (fif	in) 519	307	1.83	6.2.6	613	EIF	8.87	76.5	684	6.05
										***	6.60
	Eswat Diameter (ft)	111/4	10.68	11.6	11.6	111.67	18.85	13.75	11.6	18.5	
	Burst Volume (eable fee	1216	61.4	8:50	876	£03	87%	1370	706	886	18.85
	Assent hete from										55.3
	SF0 to 5,000	688	67.6	267	658	51.8	1005	1191	500	861	ا ا
)	5,000 10,000	330	355	365	618	608	1186	1156	754		486
	10,000 15,000	676	1.71	367	1.85	1.18	me	1086	666	(36	976
	15,000 20,000	566	686	1.83	1.68	1.78	1000	901	656	5786 - 878	193
	20,000 25,000	57.0	48.5	617	cor	656	608	783	785	686	E7G
	25,000 30,000	1.55	CIG	1,15	690	(31	760	718	666	756	766
	30,000 35,000	508	1.71	1.83	SOF	1.88	634	781	100 70 70 70 100 100	701	736
	35,000 40,000	EIE	50.8	4.59	485	511	180	760	178	884	778
	10,000 15,000	556	575	CIL	580	841	705	236	786		795
	45,000 50,000	506	686	450	550	5/.1	8.08	655	£74	560	7.5.5
	50,000 53,000	610	714	586	500	art	800	753	781	648	757
	55,000 (5,000	768	ME	686	6,25	667	814		750	664	814
	Minutes in Flight from						4 1 1 1 1	380		678	688
	500 10 5,000	7.0	9.8	18.1	6.5	6.6	1.5	4.8			7-1
	5,000 10,000	5.3	18.7	187	€. €	6.6	4.6	4.8	6.1	_5,1	4.8
	16,600 15,600	7.4	10.0	16.6	11.5	100	1.8	25	7.6	6.6	8.1
	15,000 20,000	8.6	11.6	11.0	11.3	100	8.6	6.8	76	6.8	€.7
	20,000 23,000	11060	103	188	1115	10.5	6.5		66	8.7	€.7
	25,000 30,000	16.1	9.7	18.5	16.8	105	6.5	6,4	73	6.1	6.4
	30,000 35,000	6.9	16.5	118	9.8	11.8	6.6	6.4		6.3	6.6
	35,000 40,000	9.5	8.8	110	16.3	10.6	6.1	6.7	6.5	66	6.5
	16,000 15,000	9.0	6.7	6.6	9,6	6.5	7.1		6.6	5.0	6.3
	15,000 50,000	3.3	4.1	10.8	5.1	5,8	8.6	5.8	61	66	6.6
	50,000 55,000	1.7	50	9.4	64	1 75		6.5	64	6.8	6.3
	33,600 60,000	1.8	66	7.5	5,8	6.8	8.8	6.4	117	6.6	(.1
	(6,000 6,000						1 60	5.6		8.6	8.3

-	1		il Taran			Iverage of	#30)	198	F93)	/ 100	Average of
	Augustin bi.	#197/	#19E	1120	#137	1. Flackts	10	16	10	11	
	Terricos (Ibs)	110	365	361	/1 501		700	700	710	700	
	Free Little (grans)	51.1			6180		5590	5550	5550	5550	
	(constituti (const)	5150	6181	5/80	C-16- 87			8.5-57	6-5-57	6-6-51	
	Date	() 11 E1		5:80 610	11: 3821		11:15 1/8	E:39 1.55	10:00 112	111.000/10	
	Steri	111.30 LH 18: 63 FH	7:8111				18.81/1	9: 450,00	11:15 1.15	18:46.115	
1	Finite	63	66	105	65.7	60	66	76	65	66	66
	Sinc in Fidght (min)			201/218	211/281		11/886	111/882	F17/851	215/25%	
	5, /5, (9)		1113/113	16:3/187			mistres	1118/159	1111/159	matter	
	F ₁ /F ₂ (atliffers)	unt/res	18184112	C 37 7 2 4,	1691977						
	Swat Altitude (feet)	14 166	1.1.026	54176	16.614	15,357	17,083	1.5,341	18, 361	11.1.73	46,818
	Investigation of the little		1770	518	568	607	788	656	657	206	680
	1.00								l		5.
	fairet bleneter (tr.)	11.68	11.7	18.1	11.88	111.4	11. 55	11.55	11.75	111 85	11.68
	Invest Volume Coulds for) 676	765	84.5	E15	7.67	918	661_	6.60	889	876
	Aspent Sets From				1 1	1 42 1				948	665
	570 to 5,000	667	50.5	6.60	635	2.67	11.60	576	986	100	1006
\frown	5,000 10,000	SEF	781	64.6	568	683	1085	565	633	1000	888
	30,000 35,000	1:17	SUF	1.17.0	508	201	676	ESE EN	658	678	637
	25,000 20,000	1.13	481	350	505	1,87	618	1.72	600	010	53/
7	20,000 25,000	915	736	130	610	1.50	601	800	685	SEF	500
Jagarija.	25,000 30,000	481	498	487	578	210	610	588	508	634	607
	30,000 35,000	567	400	1.55	500	507	614	465	68/7	553	505
	35,000 40,000	****	137	400	864	4.08	676	580	683	698	648
	NO'000 NO	 		150	575	505	650			640	665
	15,000 50,000								1		
_	50,000 55,000										
	55,000 65,000 65,000 65,000										
	Minutes in Flight from							4	4	4	
2,77	SF0 to 5,000	7.3	5.3	61	25	6.4	4.6	6.8	11.57	5.1	14.6
	5,000 10,000	€.6	6.4	8.8	6.8	7.8	1.6	5.6	5.9	4.5	4.6
	10,000 15,000	18.0	8.6	11.4	6.5	16.6	5.8	6.4	7.6	2.4	6.0
	15,000 20,000	12.4	1116	10.8	9.1	16.6	24	9,7	9.8	6.2	7.5
	20,000 25,000	10.0	11.4	18.8	10.6	10.7	3.3	166		8.6	8.4
7	25,000 30,000	111.8	11.6	11.6	16.6	1 11/1	7.6	8.8	6.6	7.5	€.3
(')	30,000 35,000	110.4	11.2	11.7	9.6	16.6	8.3 8.3	16.3	6.4	9.7	8.9
	35,000 70,000	9-4	11.5	18.3		9.5	7.4	5.8	7.8	6.6	7.6
	16,000 15,000		11.6			1617	3.5	1	1 "	2.3	1.6
	15,000 50,000	1	4	11.5	1.8.				1		

					Interest of						1
Legenties Sec.	139	110		19	N.FIthte	$=\pm$					
Feylock (Abs.) Feer lift (grand) Grass lift (grand)	2 8 8 8 1101 14	1000 1000	2882 1148 14	10 1000 6868							
 Date Stert Finish Finish Flight(minutes)	THE ST HEST HEST GEN	7-11-57 1:05 (1: 2:35 (1:	7-18-59 1315 (18 585 (18)	71-20+ 87 858 AM 7103 AM 68+4	56.2						
Is /In (C) P1 / P2 (militers)	milieo	ent/sec	777/502 1105/156 115,064	50, 376	26,280						
temet elittate (feet) evo, bete of like (filal	118,628	985	914	737	£67			1		1	
Most Dienster (fest) Most Volum (suide fer	11.35	18.0 981	12.0	1185	18.03 838						
expect between from	1000	1686		ine.	1110						
5,000 10,000 10,000 15,000 15,000 20,000 20,000 25,000	1160 980 987 726	1318 1110 1318	1318	7181 981 753 663 688	1116 1633 843 755				Ť		
25,000 \$0,000	806	505	798	686	697		l				
\$3,000 \$3,000 \$3,000 \$6,000 \$6,000 \$5,000 \$6,000 \$50,000	7,30 8.16 876	788	665 685 689	616 667 617	717 737 G84						
Negatives in Filest from ST to 5,000 5,000 10,000	1.6	9,6 3.6 3.6	4.7	4.3 4.2 5.1	#1.3 #1.0 #1.3						
10,000 15,000 15,000 20,000	5.4	1.5	3.6	6.3	6.8 5.9						
25,000 25,000 25,000 35,000	6.8	5.9	6.3	8.5	6.6						
30,000 35,000 35,000 40,000	6.6	6.8	7.6	7.3 8.2	7.2						
25,000 25,000 25,000 50,000	3.8	1.5	7.3	9.5 8.5	68 1.8		1	4		4	4

Jeneralm In.	#107c	_/xxe	1380	# 12)	1 388	1125	#186	1/1820	Average of E-Plights
Feylood (145.) Free lift (great) Greek lift (great)	300 300 300	3 200 3	3 200 3	3 200 1688	3 200 MU	3 200 10 6 6	3 200 1685	3 200 1688	
Date Start Finded: Time in Pight (minutes	8-9-57 1306 AM 2:31 IM 86 E	8-9-57 3:06F/A 4:24F/M 6!!	10:25 1.18	E-12-F7 1E116 11M 11:23 FM 66.7	1:58 61	11:168 1.1%		G+15+50 10:20 11:50 50	83.6
72 / 73 (47) F ₃ / F ₂ (61113ters)	888/368 1003/85E	250/703 1005/267	737/25P 1114/257	200/288 1018 386	286/283 1008/288	233/345 1013/875	24.0/3.05 1016/214		
Singl Attitude (ft.) eve. Lete of Rice(ft./mir.	27,618 676	14,183	3 6, 18 6 336	26,811 133	36,666	34,087	30,579 414	59,149 413	30,103
First Denoter (de.,) Hirst Volume (conto fer	628) 171	159 159	100 600	75.5 131	88.0 178	80.4 189	77.2 141	618.72 1734	80.4 188
Fate of Eige (ft.//stn.) \$50 to 5,000 5,000 10,000 10,000 25,000 20,000 25,000 20,000 25,000 30,000 35,000 35,000 40,000 10,000 45,000	\$27 373 374 440 440 440 472 472 500	FINITY TRACES	373 250 237 367 367 486 486	386 393 478 478 478 478	466 368 348 367 413 483 487 37.0	36-6 38-7 270 397 398 489 683	484 456 388 373 380 416 414	386 356 373 381 487 468 438 757	989 386 375 388 388 387 493 489
No. 4n F18324 (ntn) SR 4n 5,000 5,000 10,000 10,000 15,000 15,000 20,000	9.11 13.41 13.3 14.5		18.7 14.3 14.8 18.8	18:4 18:7 11:7 11:7	16.3 18.7 18.6 18.6	18.6 10.6 13.5 12.6	9.7/ 111.5 112.5 13.4	18-4 18-8 18-6 18-6	11.2 12.8 13.3 12.8
20,000 25,000 25,000 30,000 30,000 35,000 35,000 40,000 40,000 45,000	12.E 11.A 10.6 41.0		10.2 11.0 12.0	168 74	18:11 118:16 11:37 (5:11	18.8 1146 8.4	18-8) 18-0 18-4	11:3 11:3 11:4 14:5	12.3 111.3 51.5